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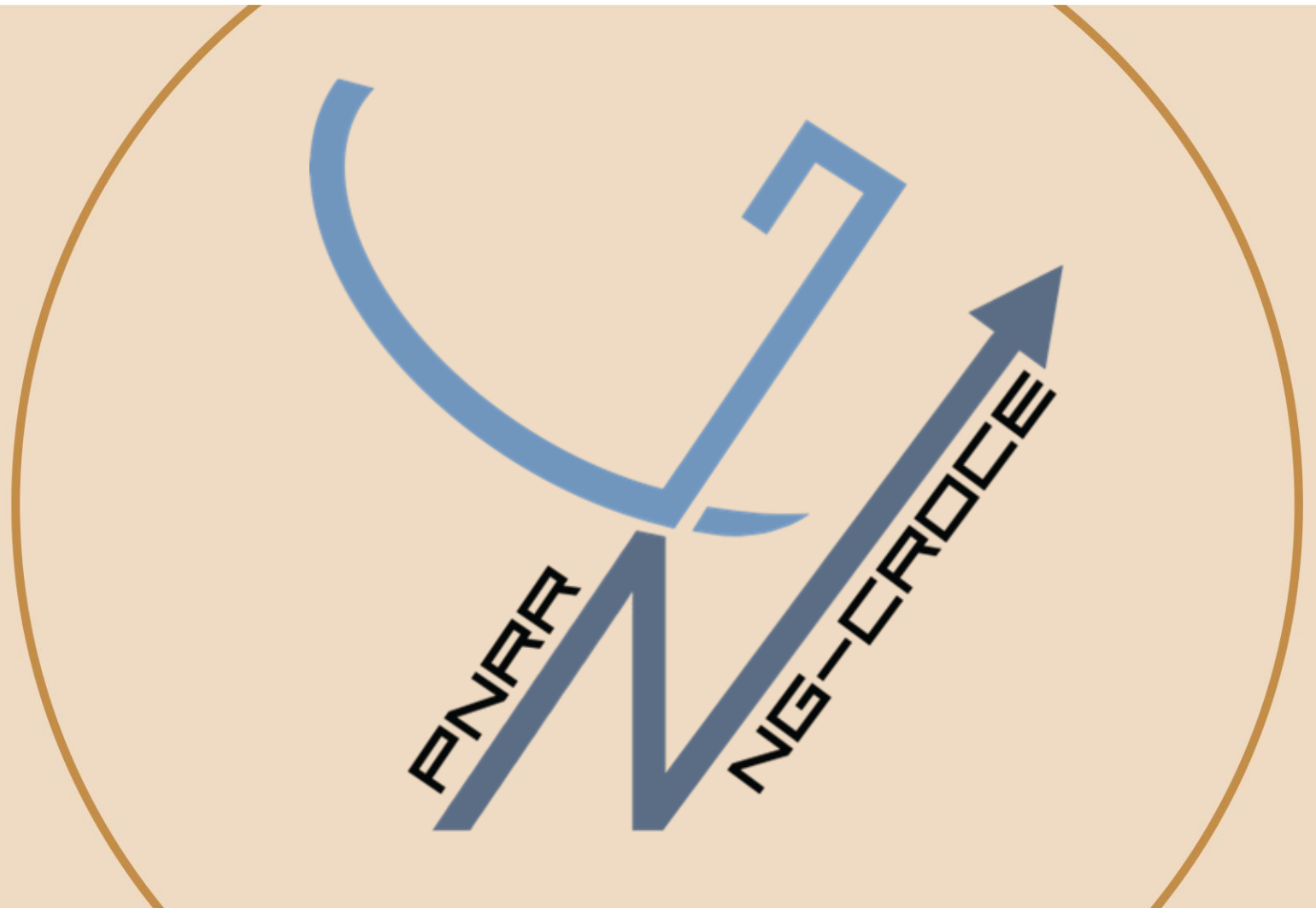
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Space cataloguing and
intelligence: the potential
disclosed by MATER
algorithm and by the East-
West branch

Final Meeting

Martedì 30 Giugno – Giovedì 02 Luglio

OAC - Osservatorio Astrofisico di Arcetri



M.F. Montaruli



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OUTLINE

- 01 MATER algorithm
- 02 Operational implementation
- 03 Ambiguity solving through East-West branch
- 04 Radar IOD
- 05 Fragmentation campaigns (CZ-6A, Starlink 34343, ZQ-2E R/B)
- 06 Conclusions



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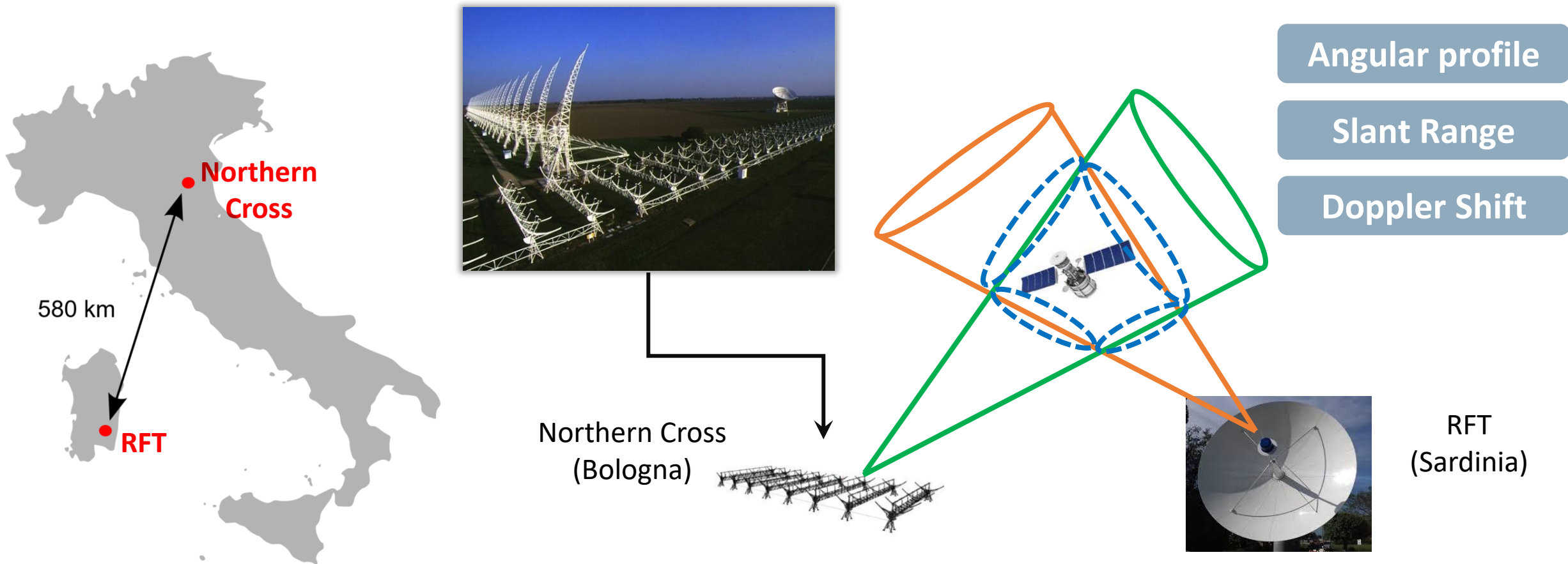


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Blstatic Radar for Leo Survey (BIRALES)





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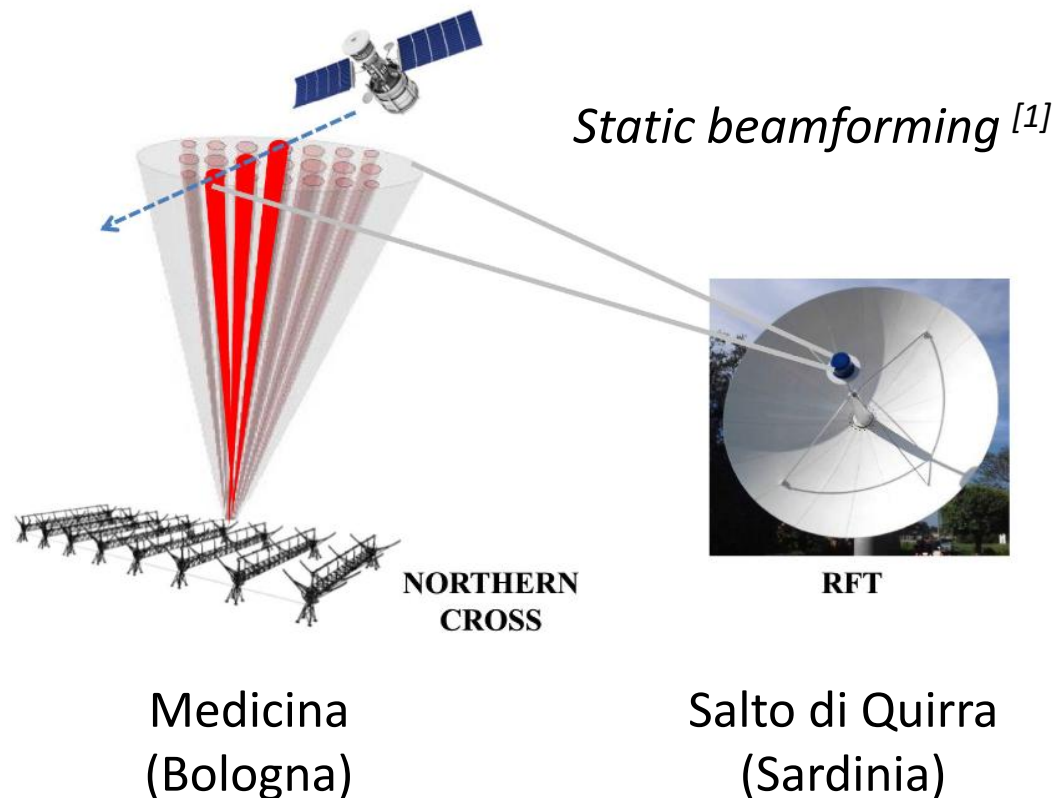


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BIRALES: Multibeam Approach



Angular profile

Disadvantage: both main lobes and grating lobes appear in sensor FoV

Beam 2



Beam 1



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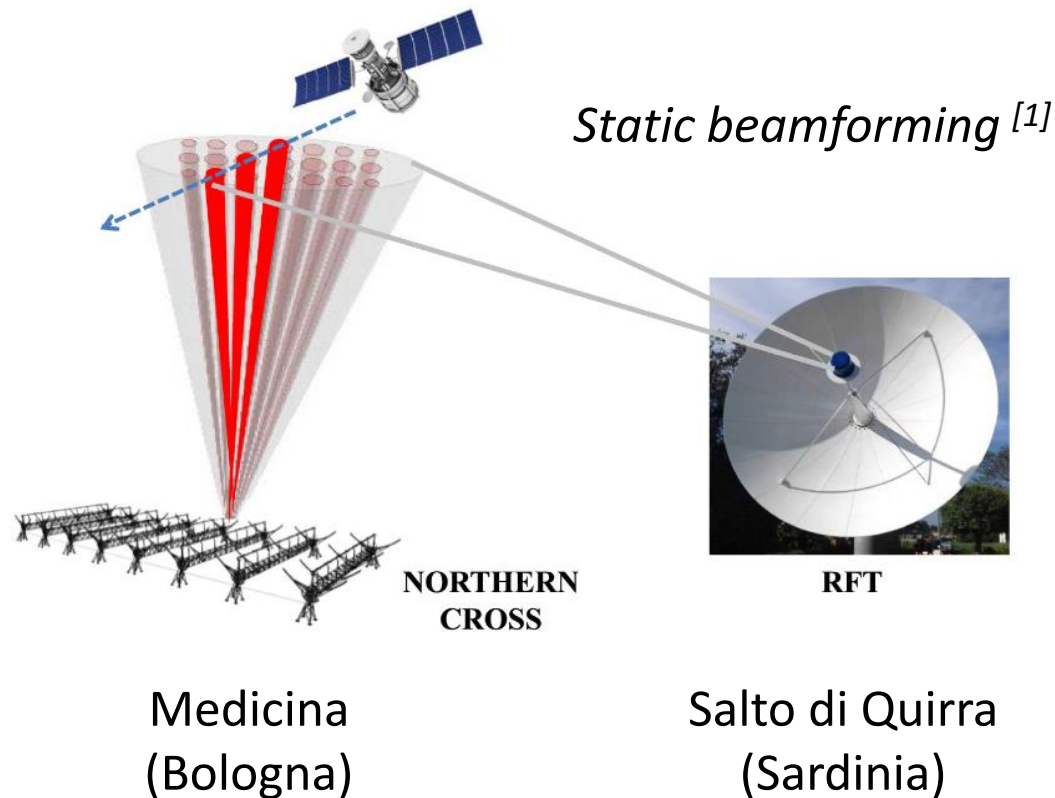


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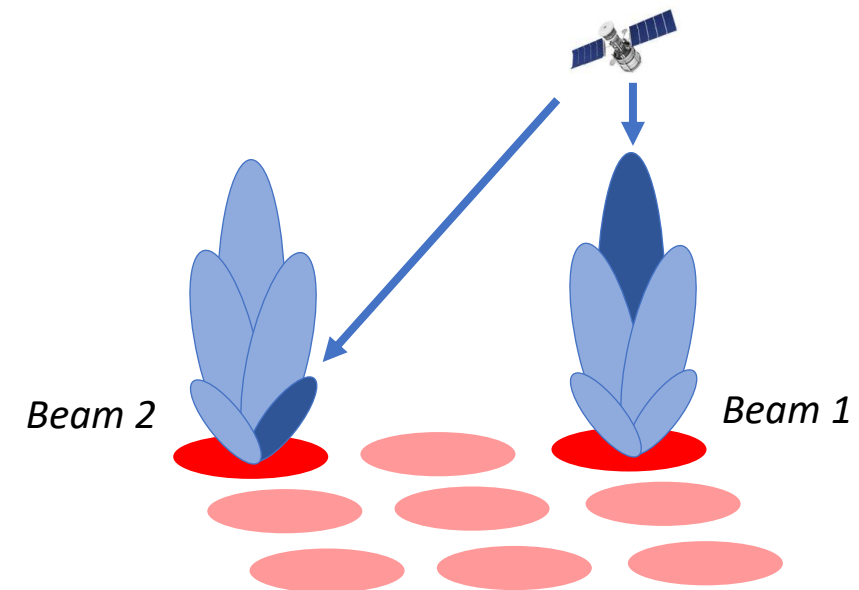
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BIRALES: Multibeam Approach



Angular profile

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M. Losacco, et al., Initial orbit determination with the multibeam radar sensor BIRALES, 2020



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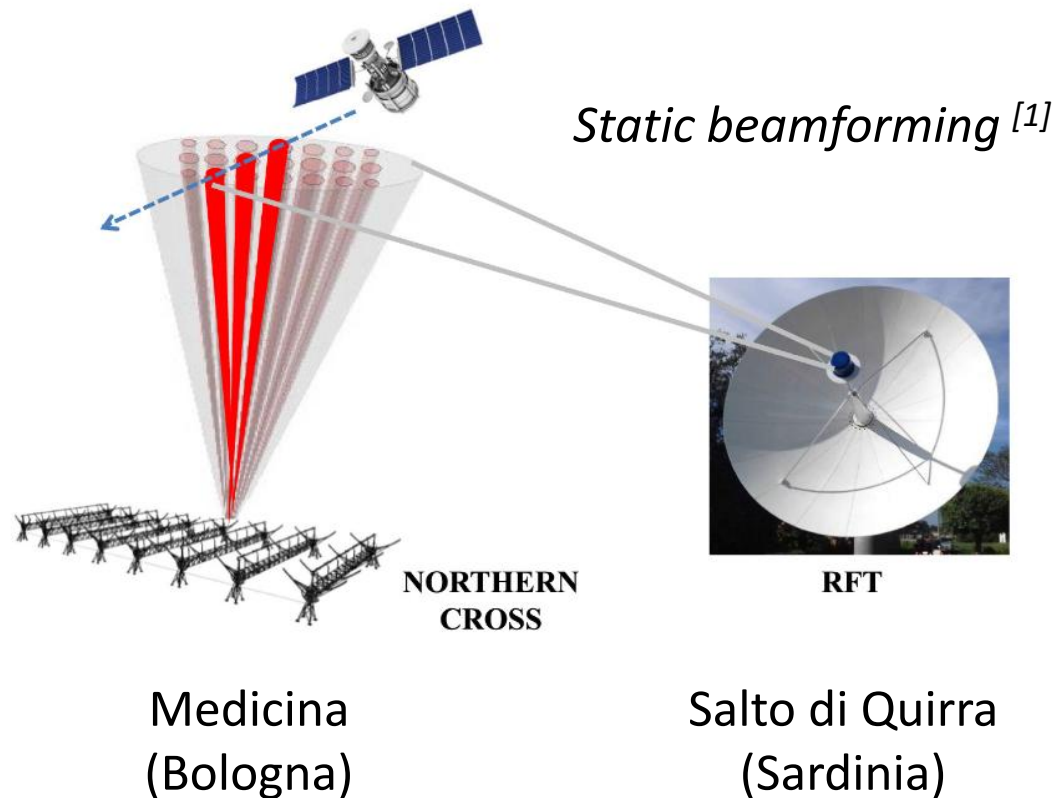


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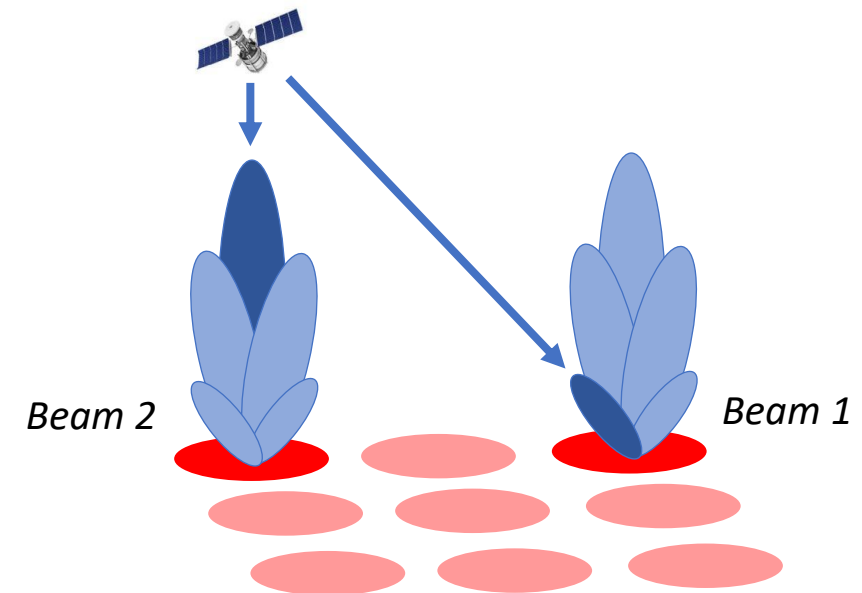
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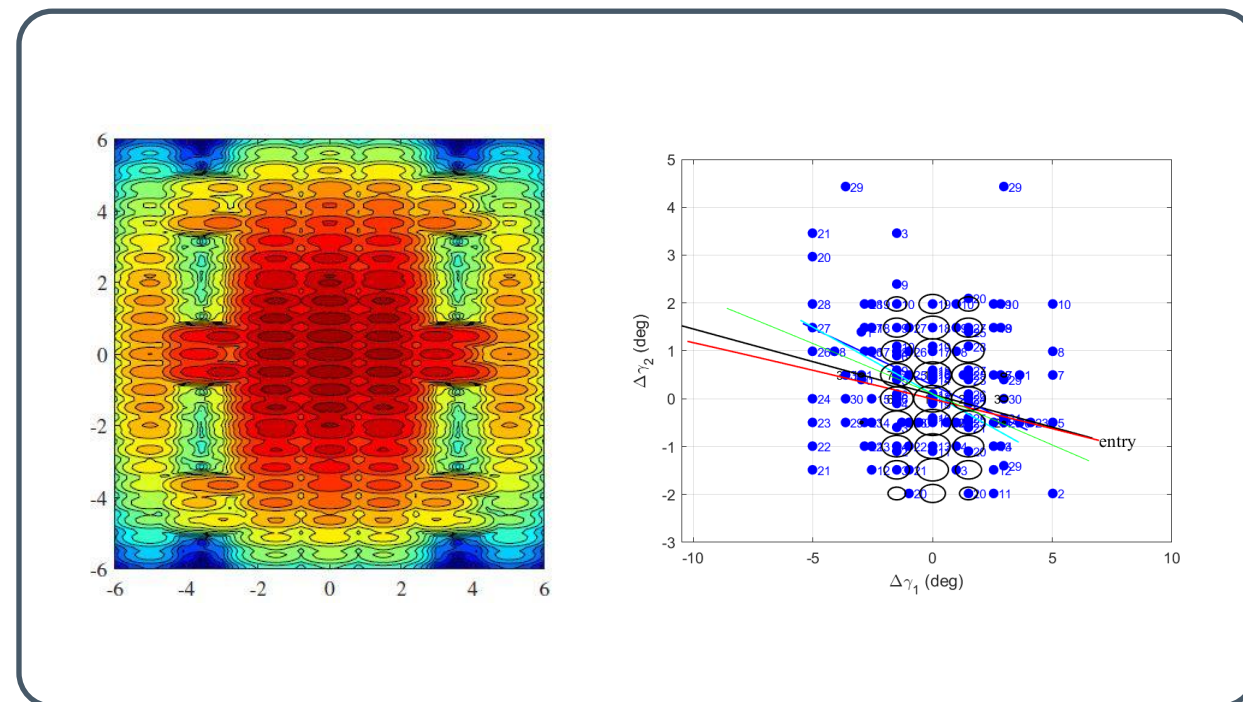
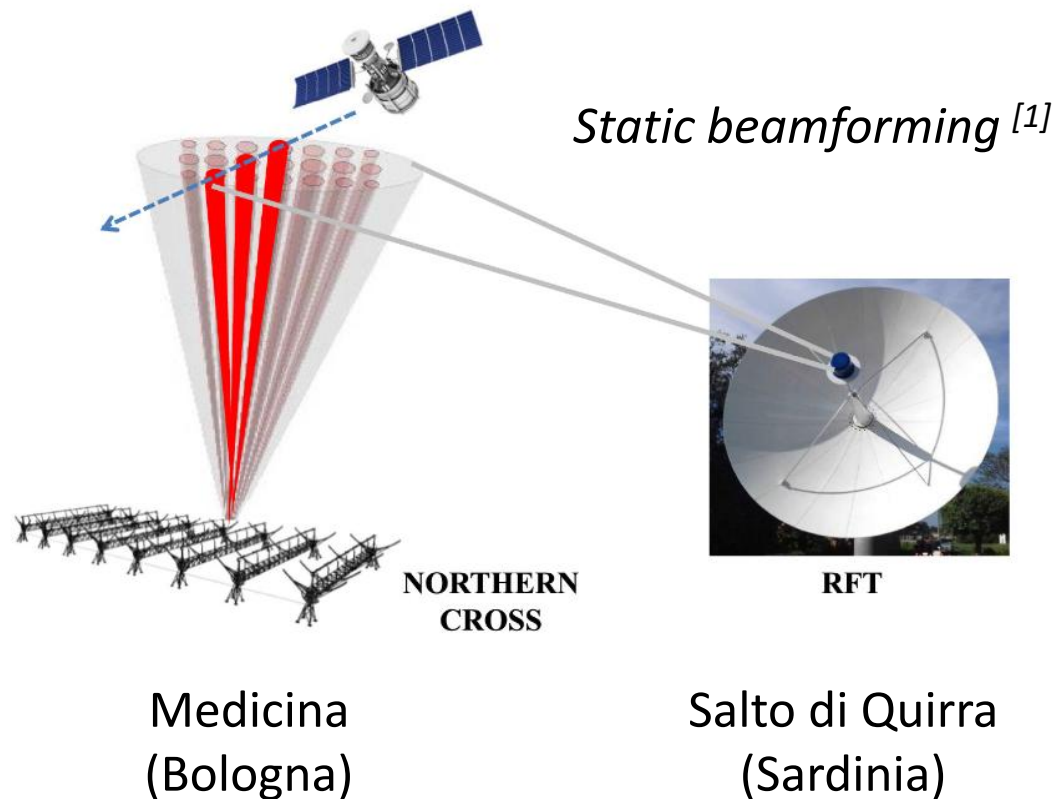


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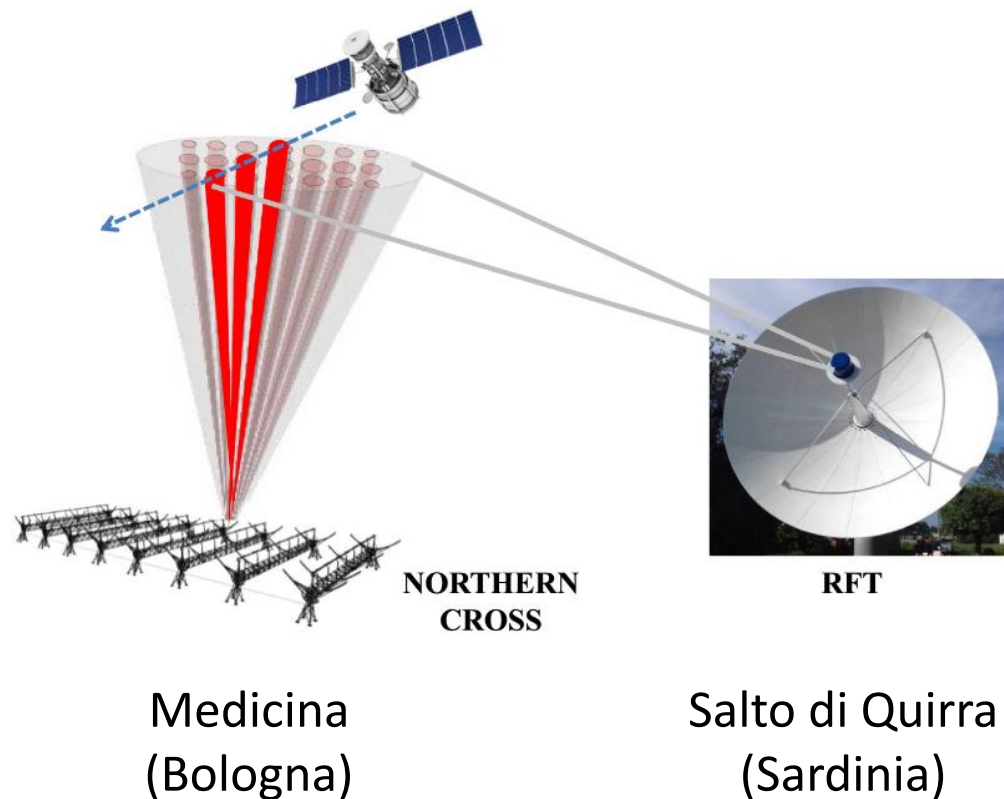


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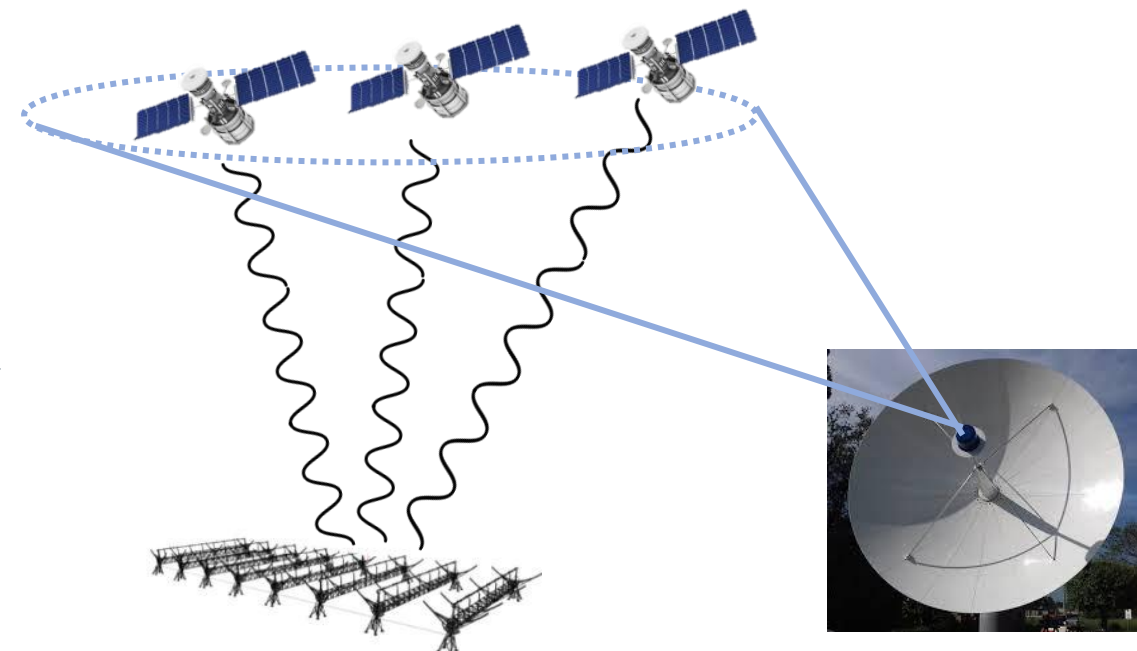


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BIRALES: Multibeam Approach



Adaptive beamforming

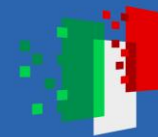




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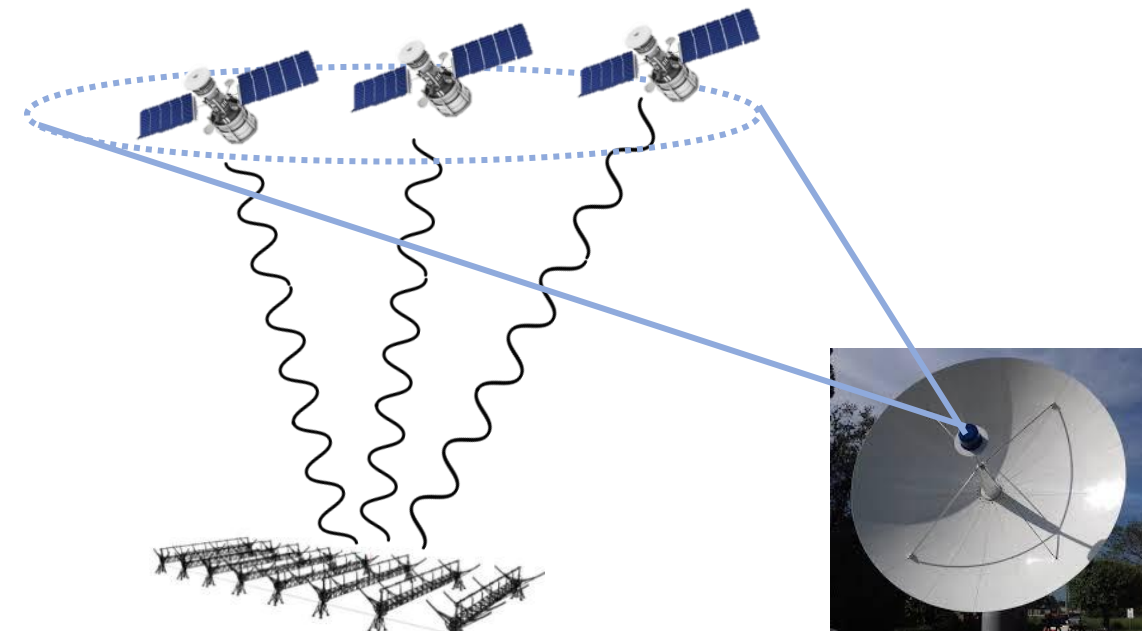


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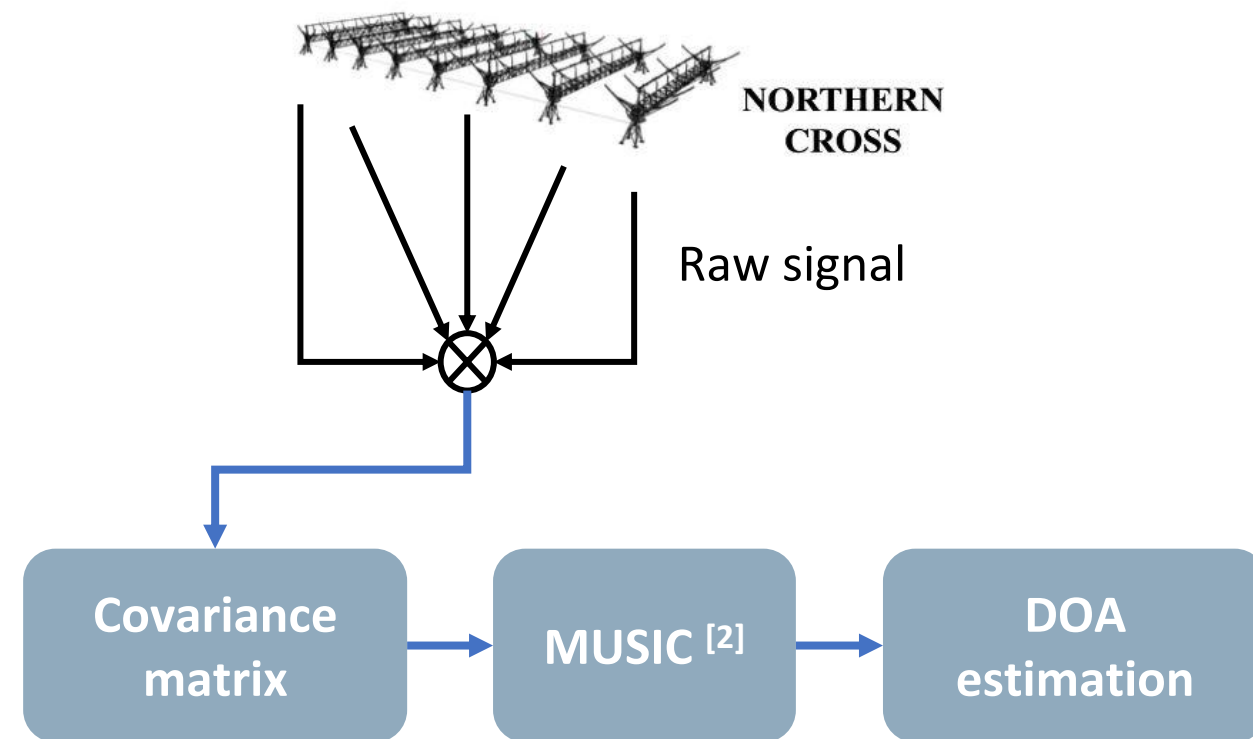


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BIRALES: Adaptive Beamforming Approach



MUSIC - Multiple Signal Classification ^[2]



R. Schmidt et al., IEEE Transactions on Antennas and Propagation, 1986



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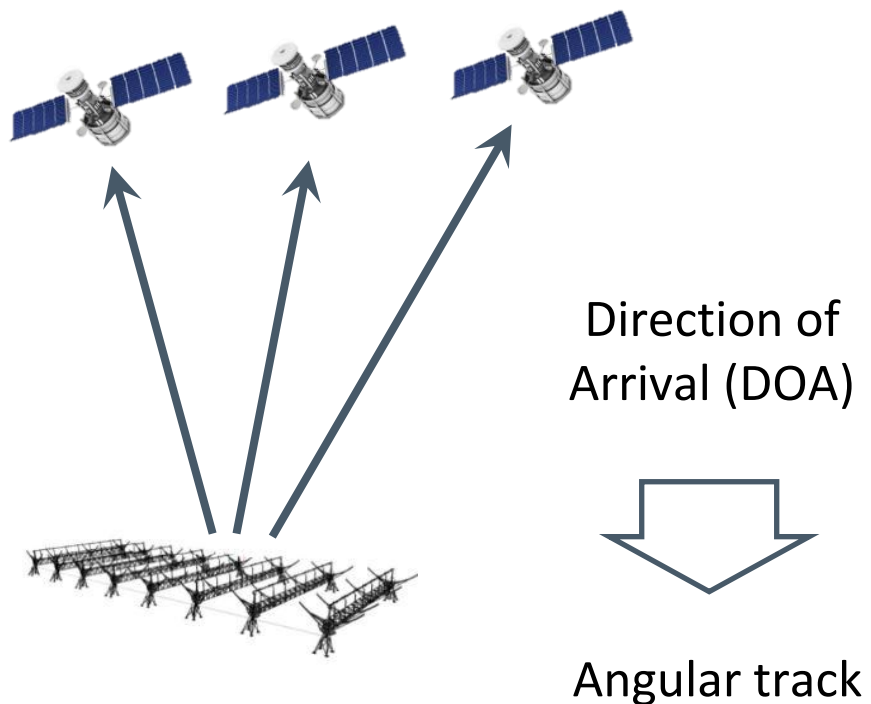


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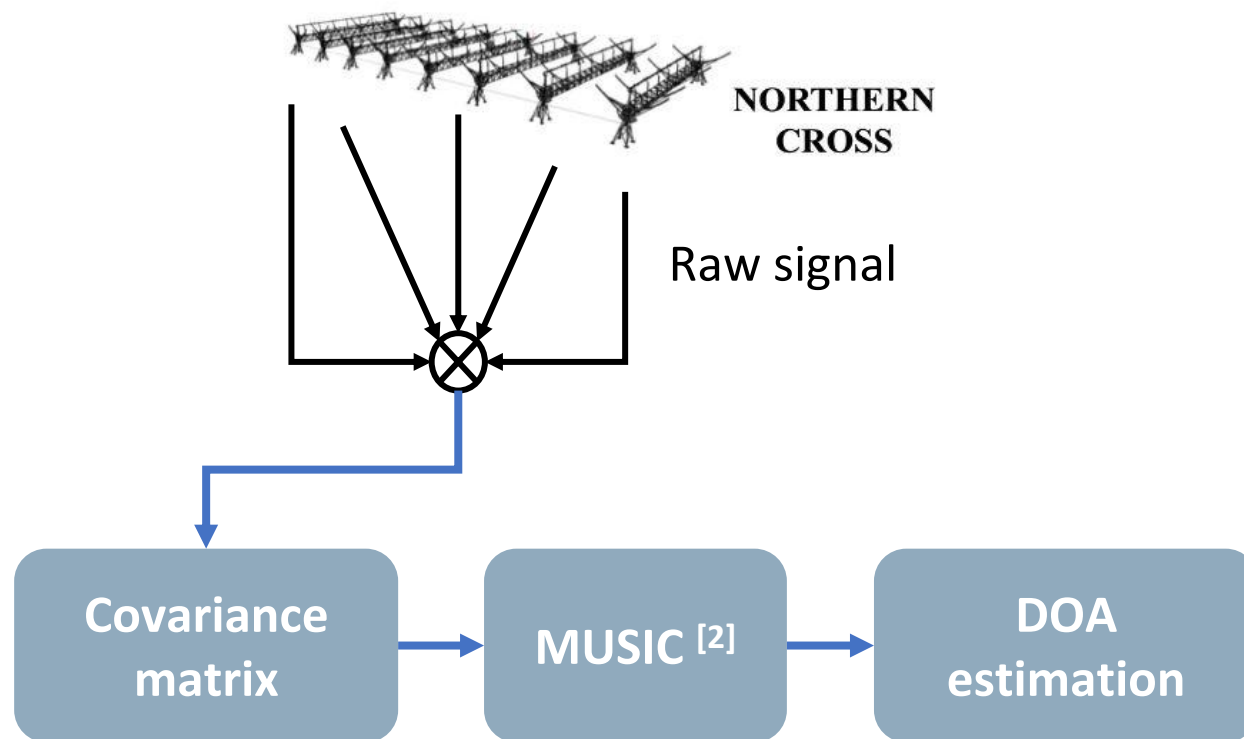


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BIRALES: Adaptive Beamforming Approach



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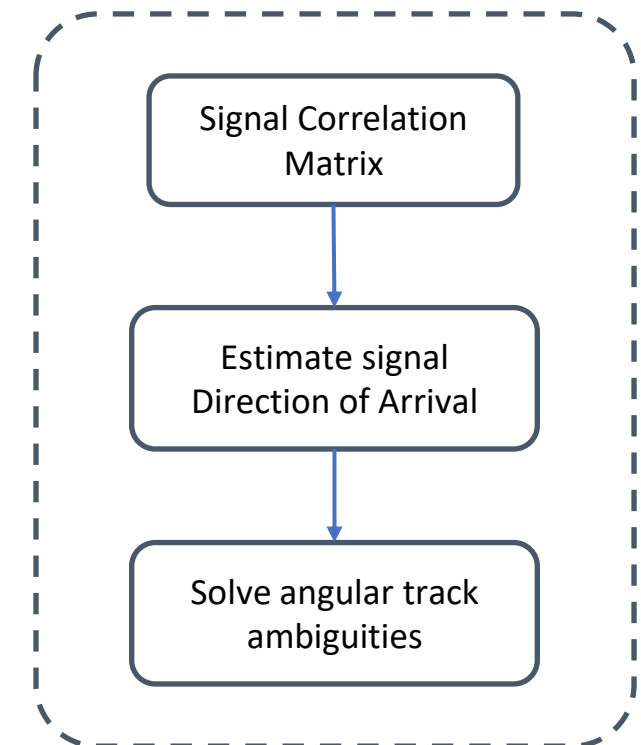
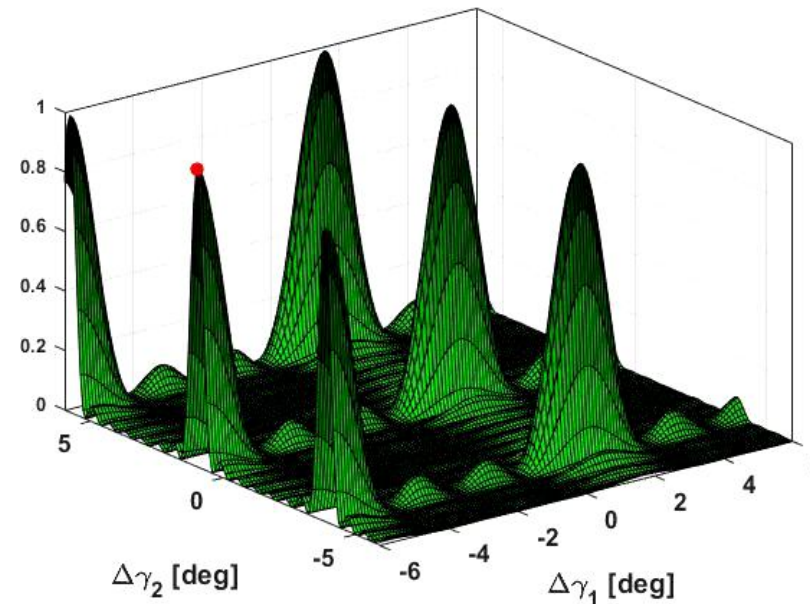
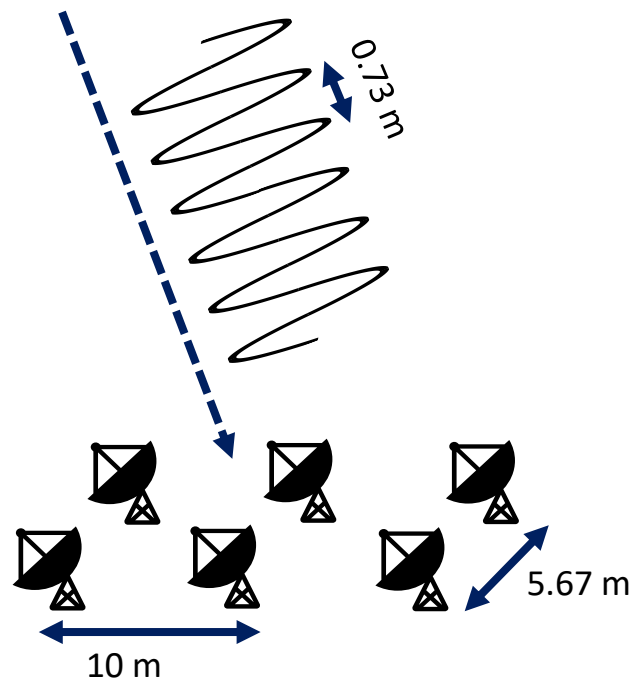


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Music Approach for Track Estimate and Refinement (MATER)



M.F. Montaruli, et al., Adaptive track estimation on a radar array system for space surveillance, *Acta Astronautica*, 2022

M.F. Montaruli, et al., Adaptive track approach for multiple sources scenario in space surveillance applications, *Aerospace Science and Technologies*, 2024

M.F. Montaruli, et al., Delta-k approach for space surveillance multireceiver radars. *Astrodynamic*s, 2024



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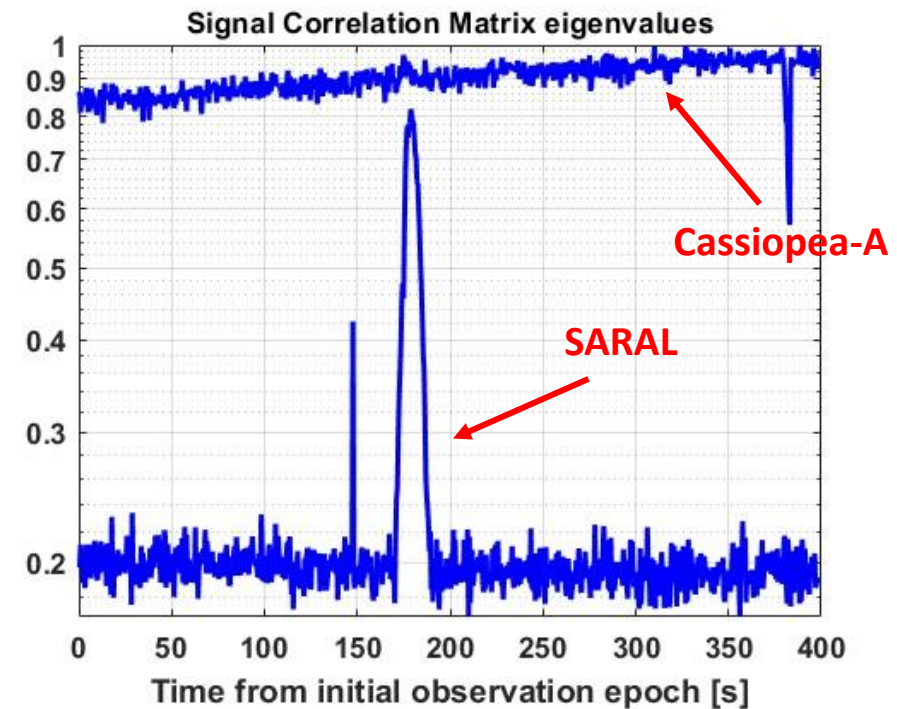
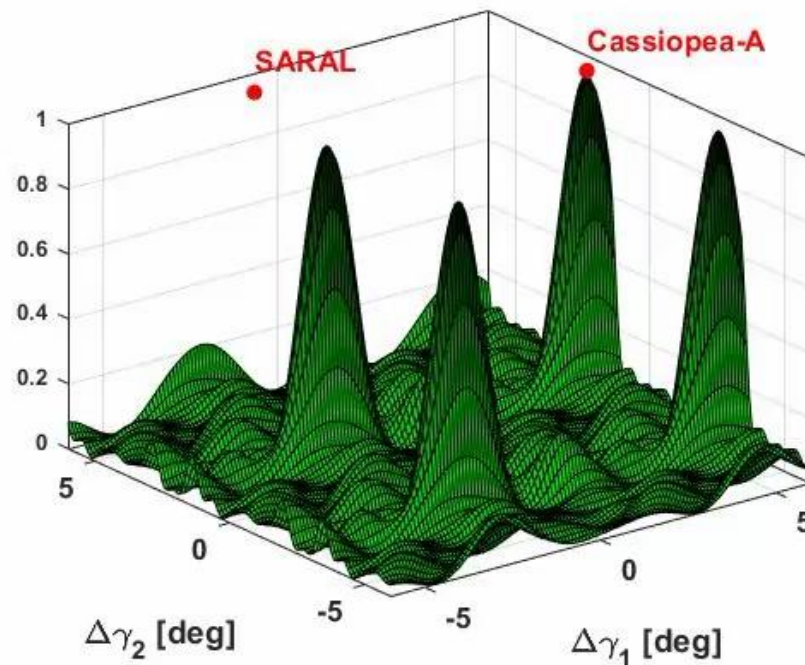


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Real data processing - channelisation

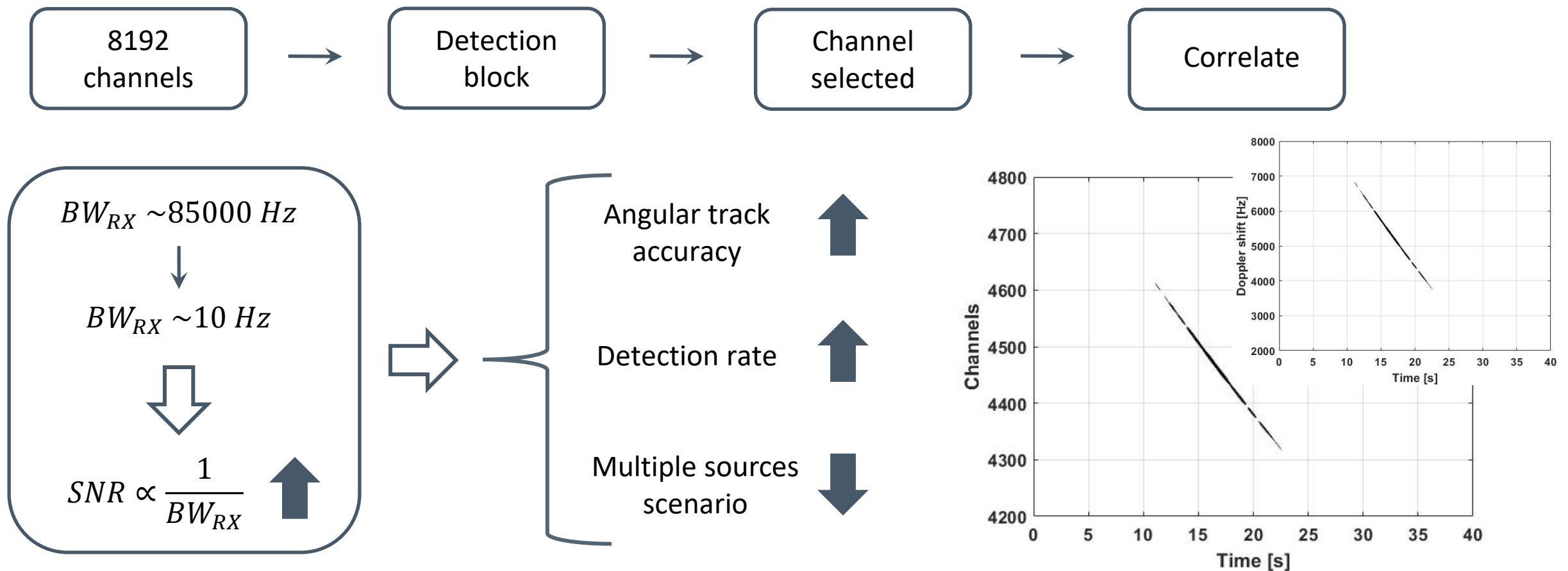
SARAL transit

- ▶ December 2°, 2022
- ▶ Target: SARAL (norad ID 39086)
- ▶ Radiosource: Cassiopea-A





Real data processing - channelisation





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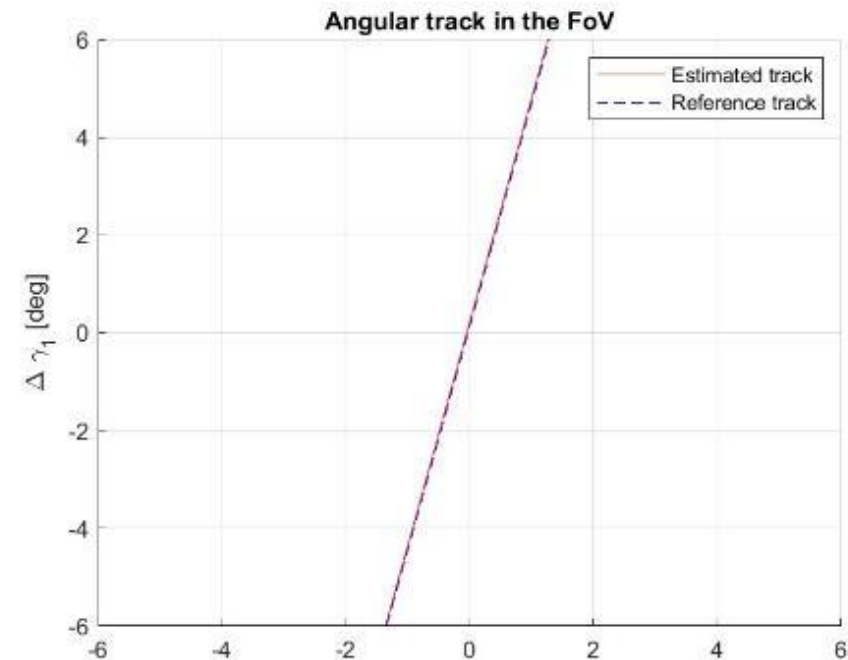
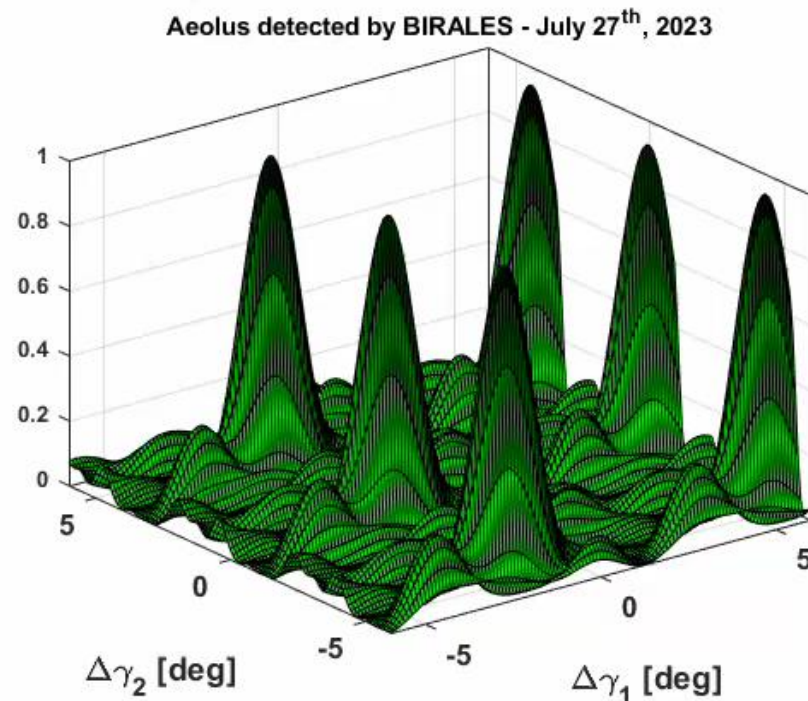


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Real data processing - channelisation

Aeolus re-entry campaign

- ▶ July 24°-28°, 2023
- ▶ The target was maneuvering during the observation
- ▶ Uncatalogued case





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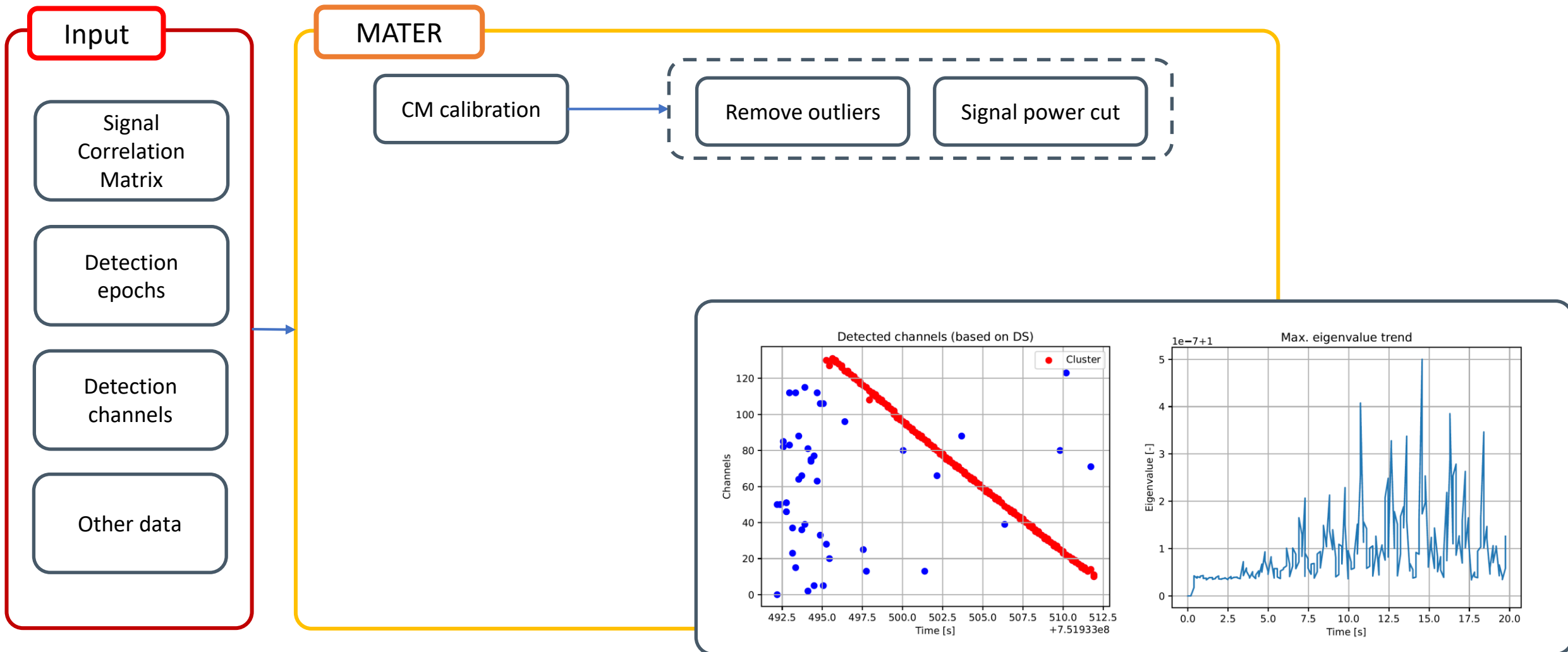
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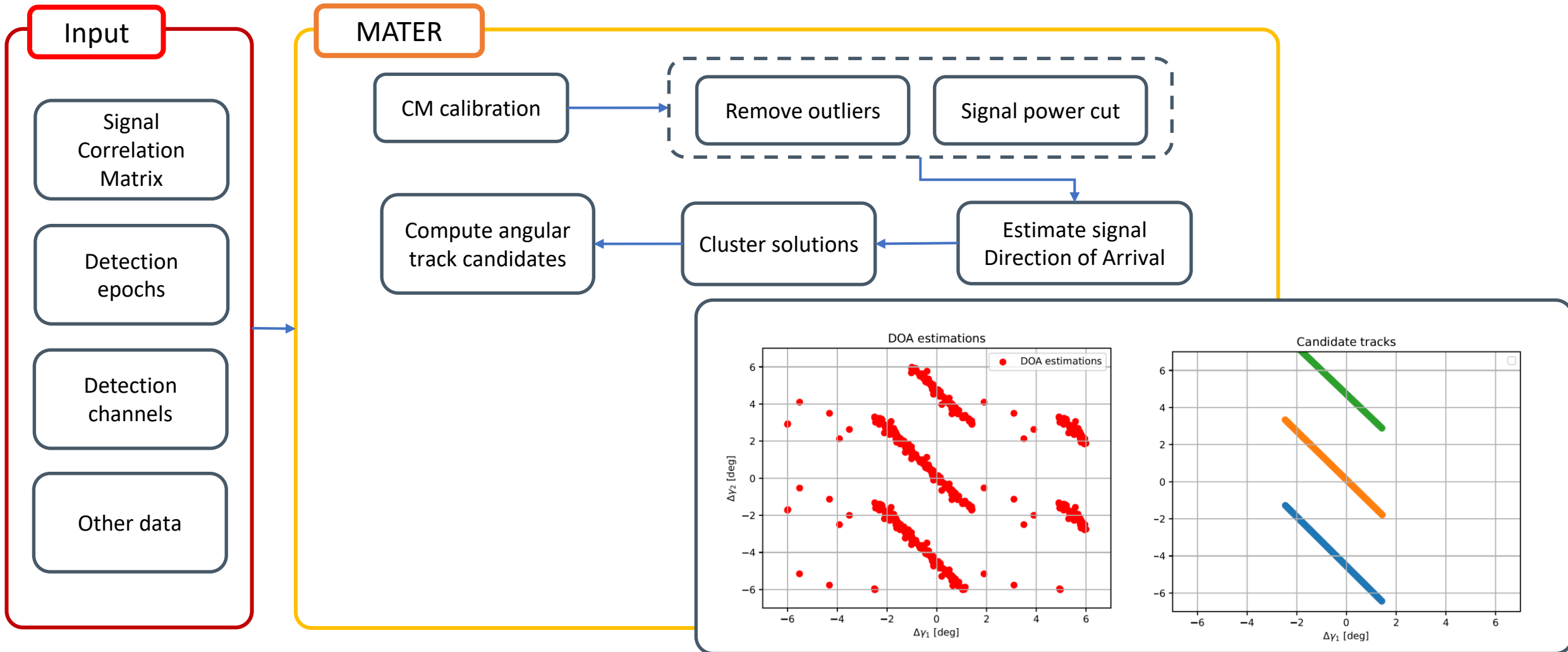
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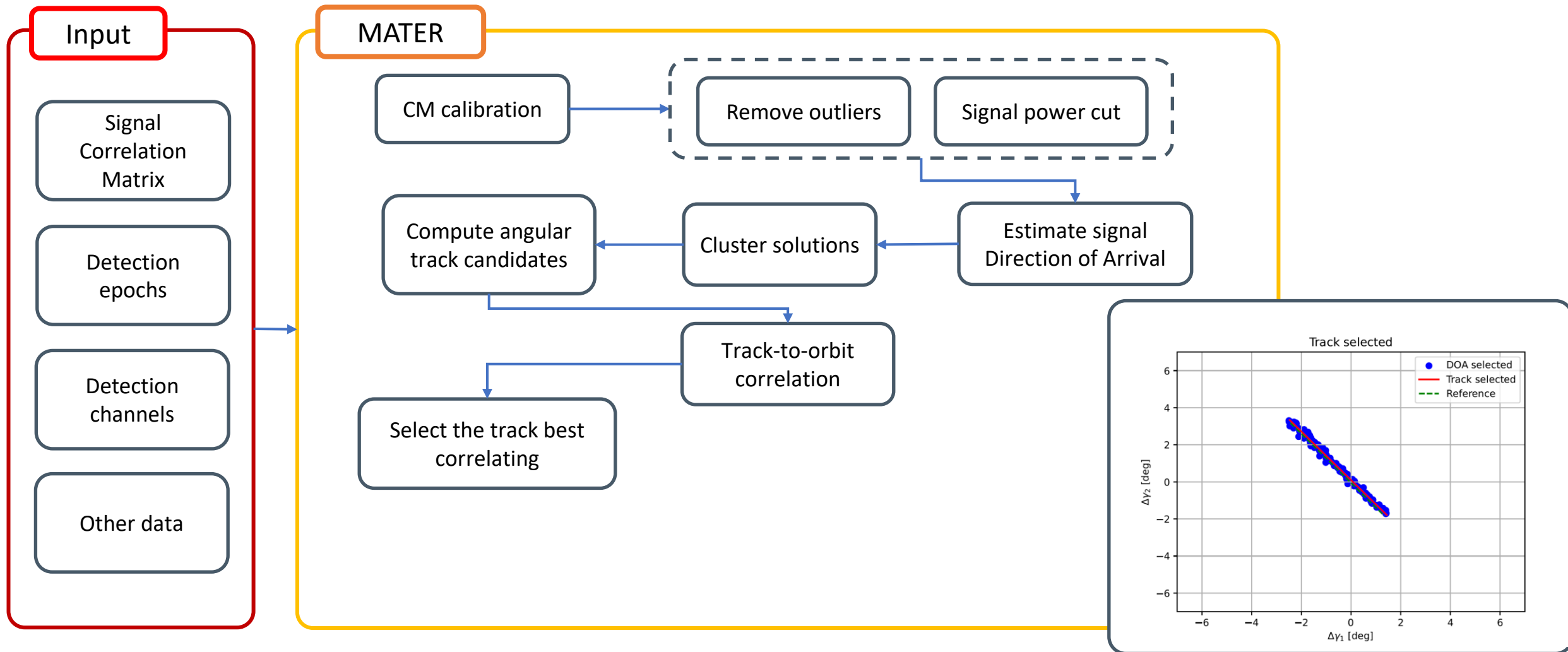


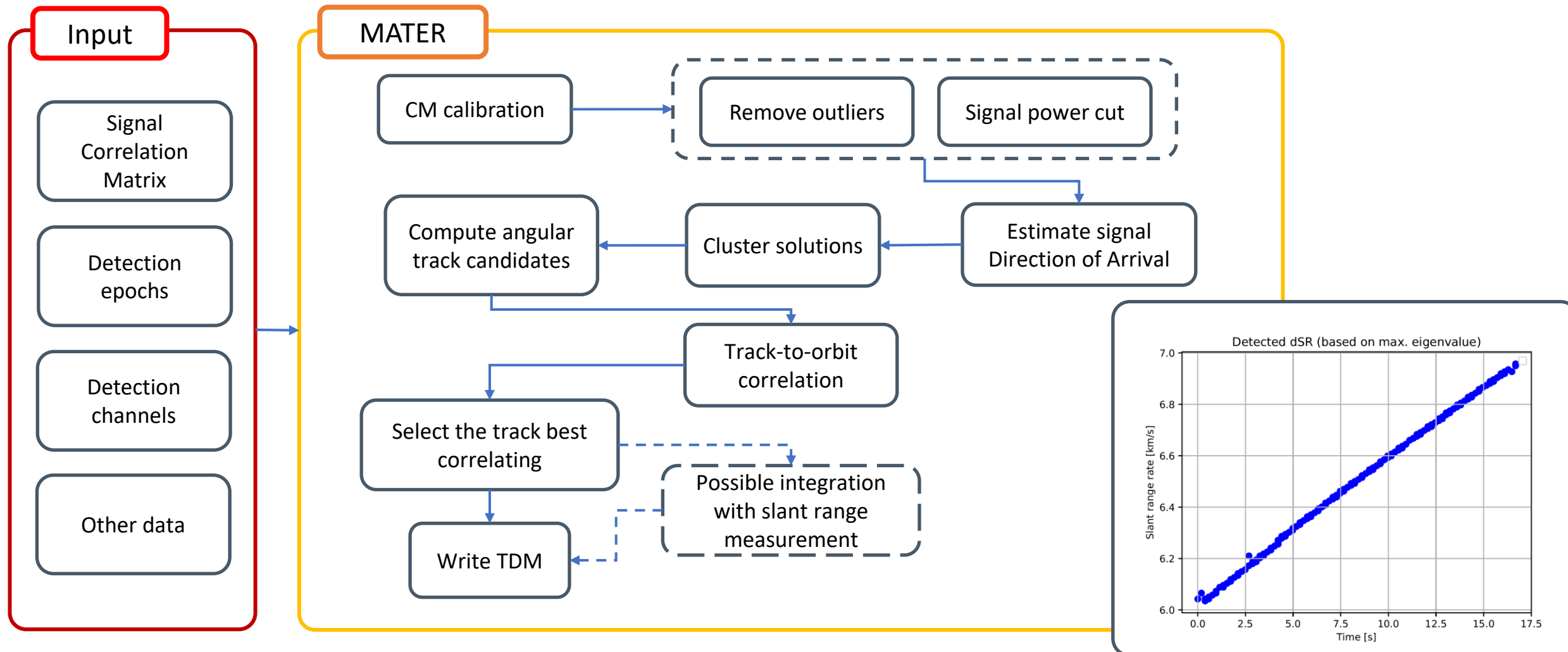
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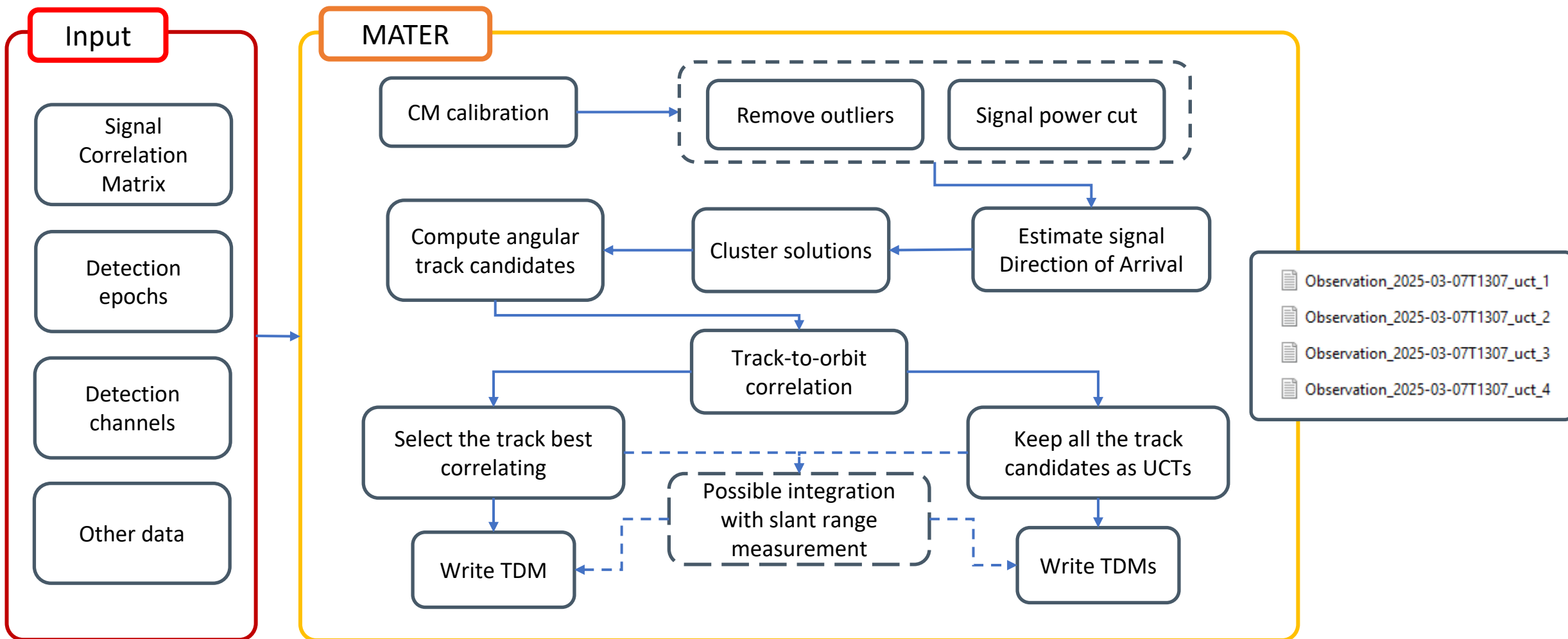


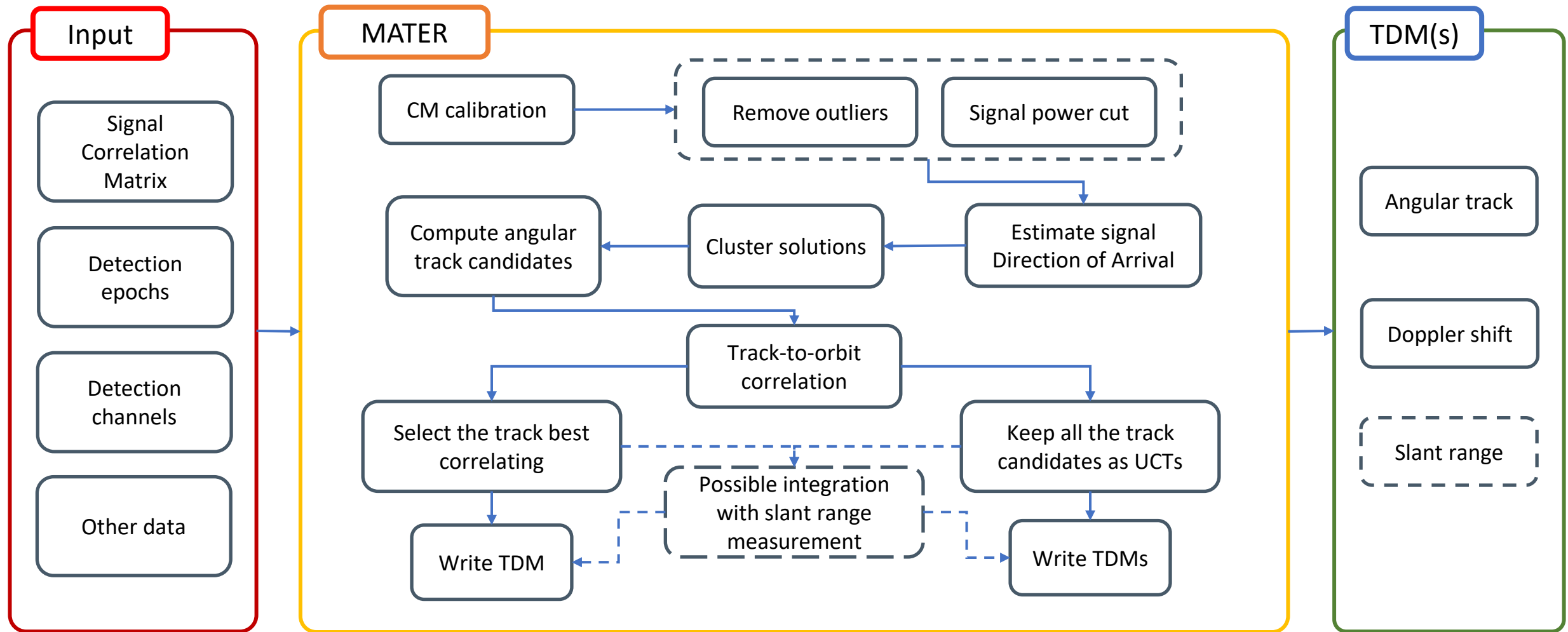
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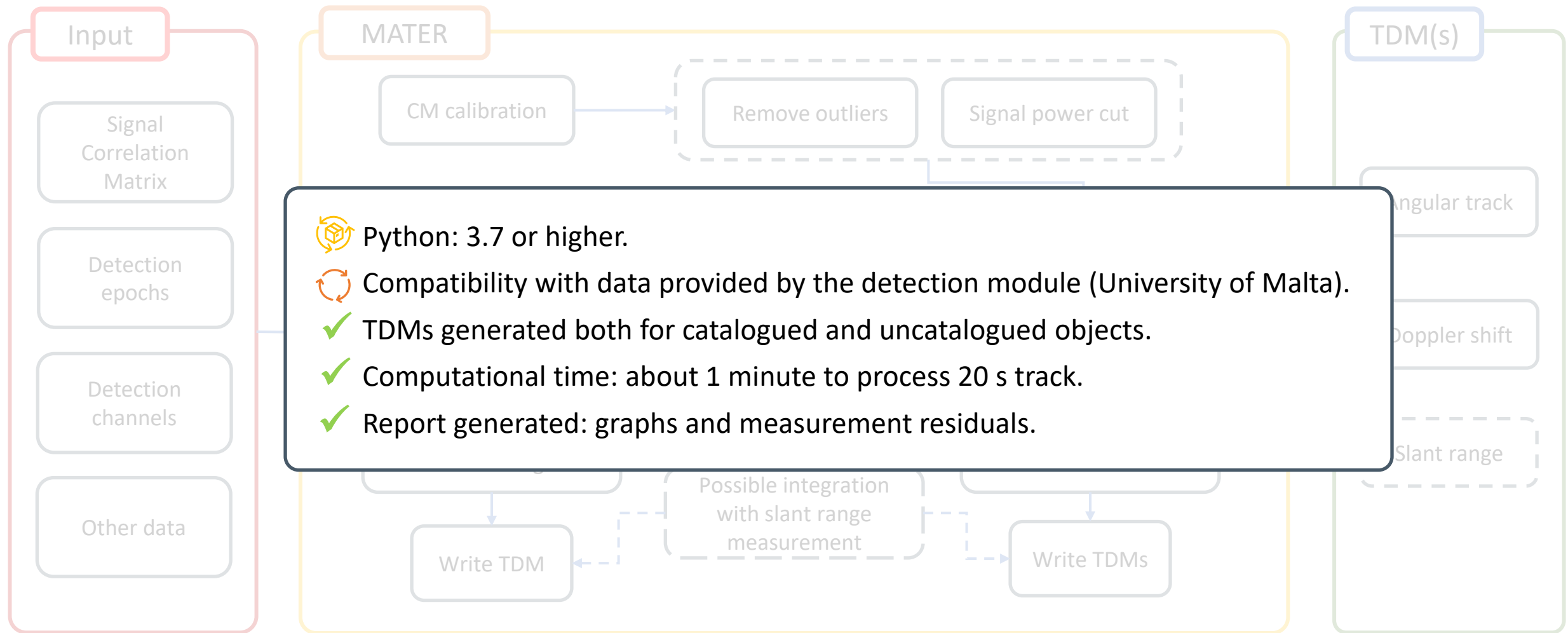
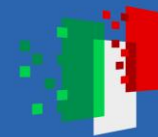














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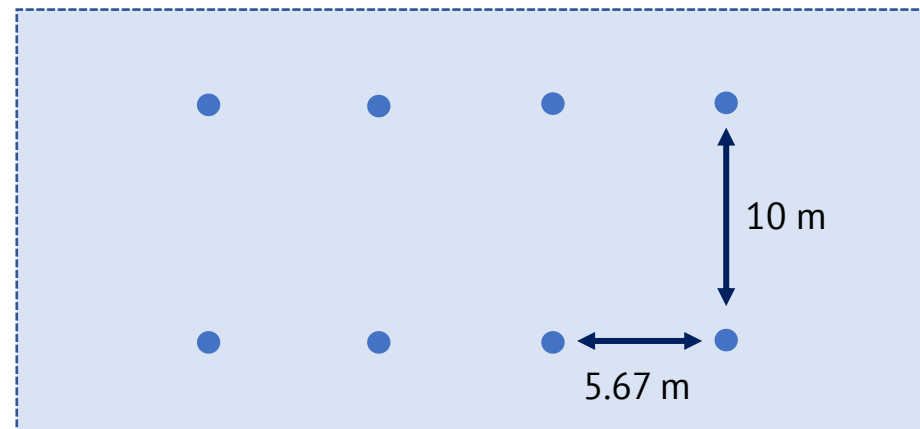
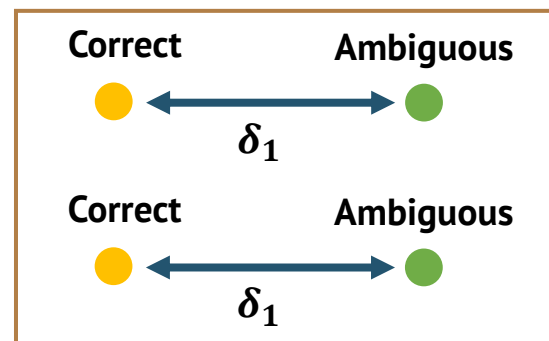
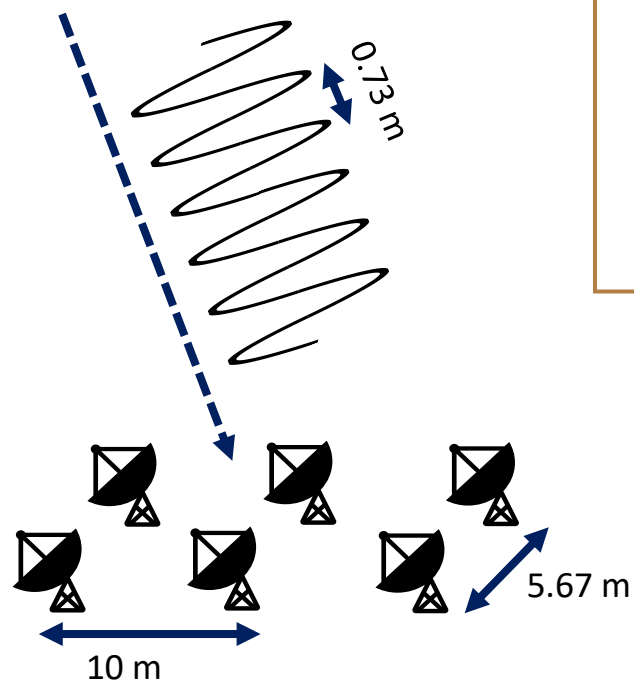


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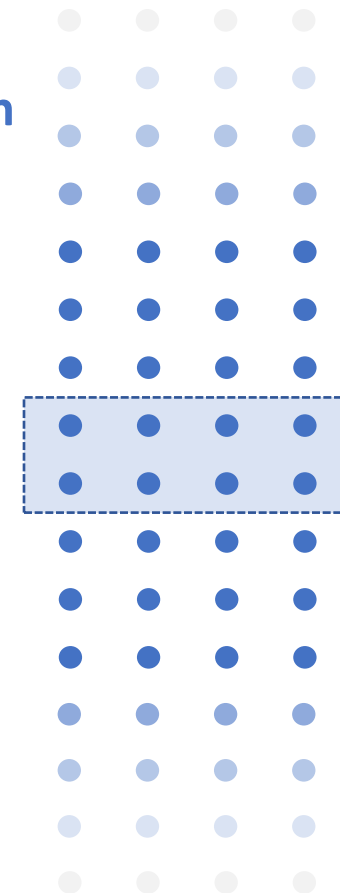


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Ambiguity solving through East-West branch



North-south branch





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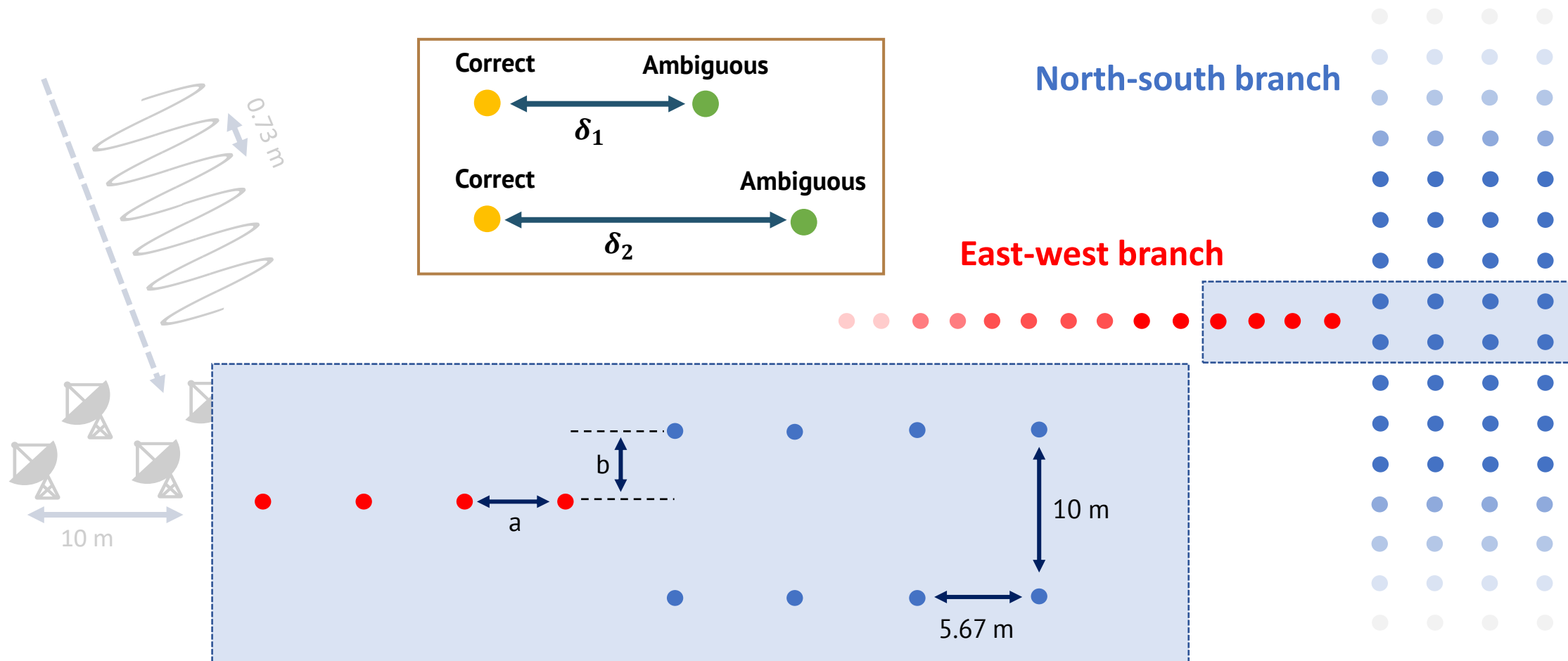


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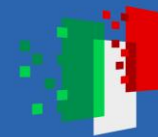




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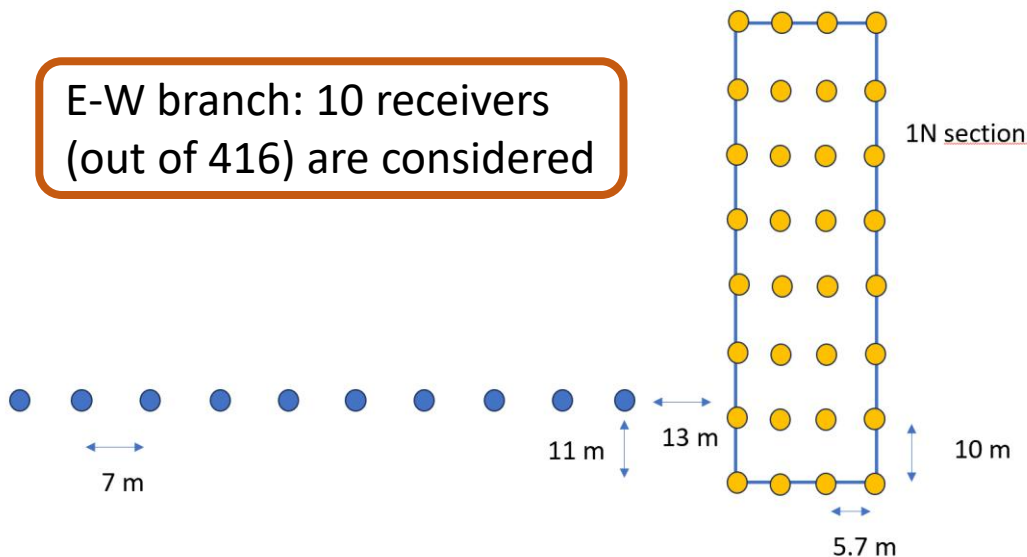
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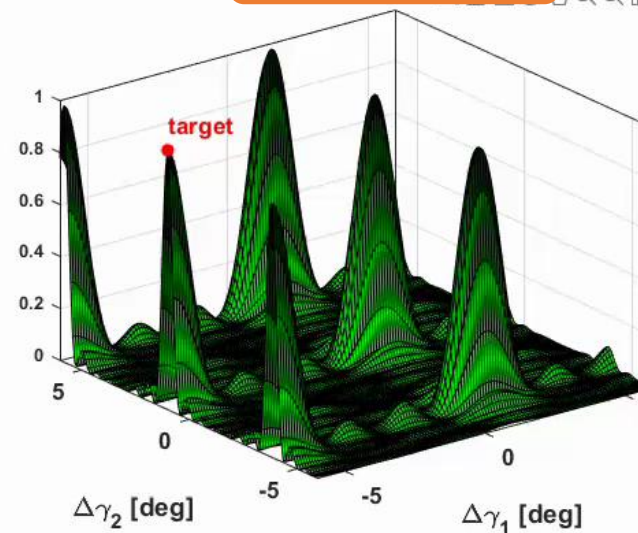
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Ambiguity solving through East-West branch

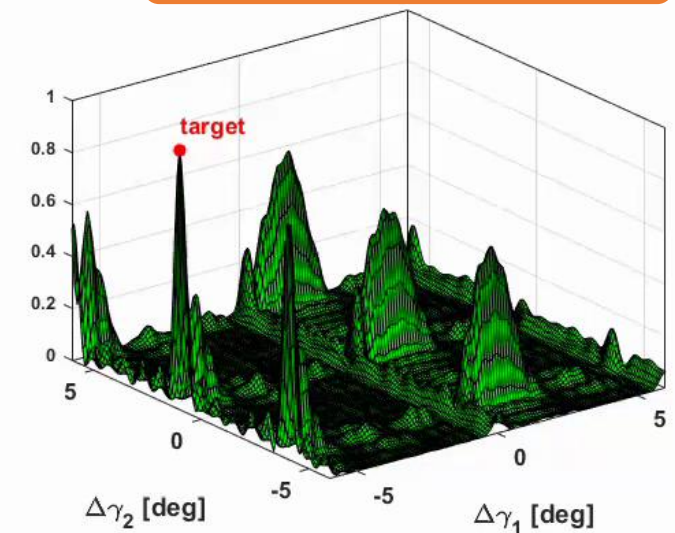
E-W branch: 10 receivers
(out of 416) are considered



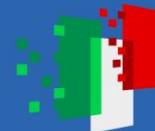
1N section



1N section + EW branch

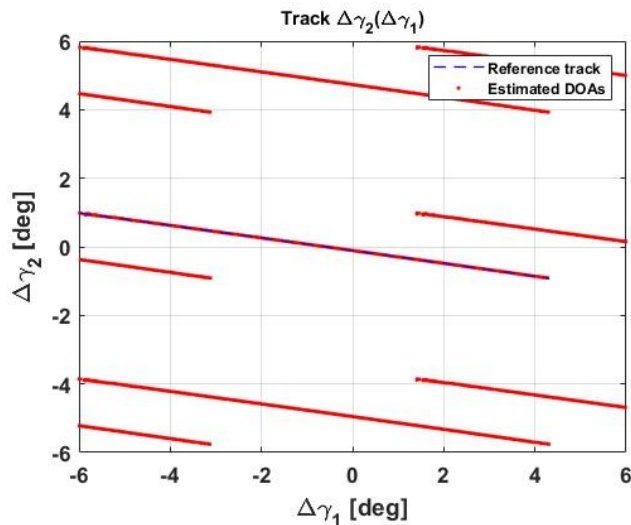


- Improved resolution in E-W direction.
- The N-S shift between the 1N section and the E-W branch depends on the E-W pointing direction.
 - Performance partially change

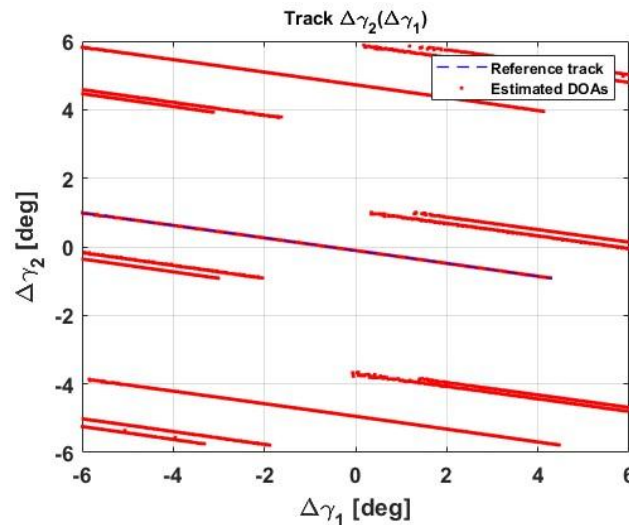


Ambiguity solving through East-West branch

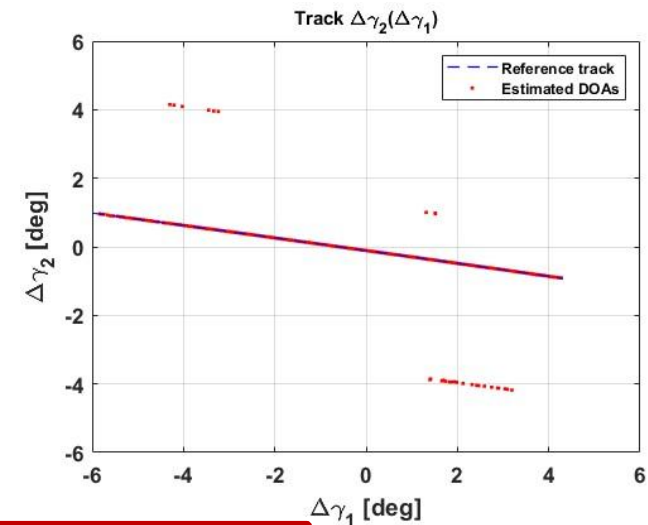
1N section



1N section
EW branch



Most fixed
solution



- 🔄 The algorithm is tested against a data set of LEO 899 transits.
- ✅ The ambiguity resulted to be solved in 100% of cases (98% with 0 m shift).
- ✅ RMSE: $\Delta\gamma_1$: 6.9e-03 deg; $\Delta\gamma_2$: 1.1e-02 deg.

Operationally

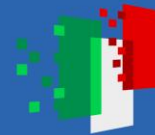
- No need of track-to-orbit correlation:
- Improves reliability.
 - Solve the ambiguity in the UCT case.



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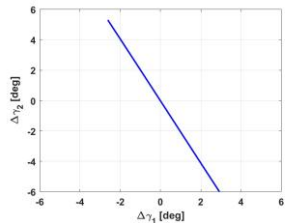


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IOD angles and DS



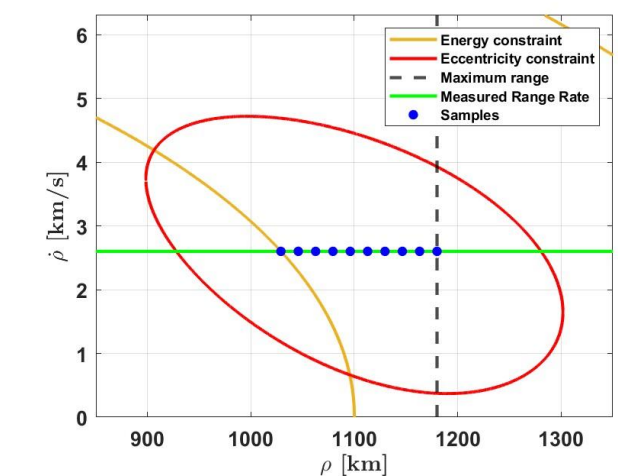
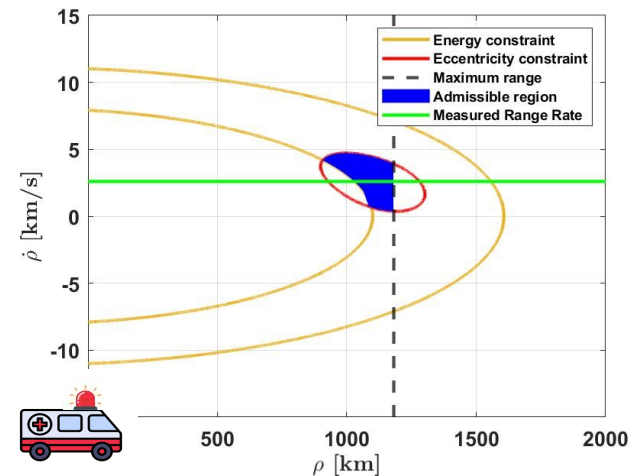
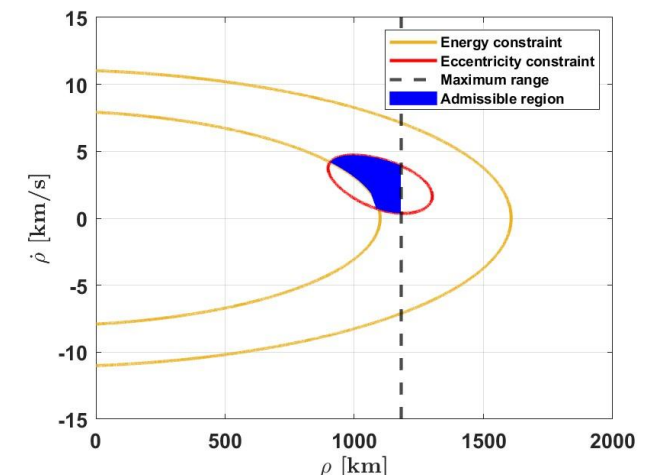
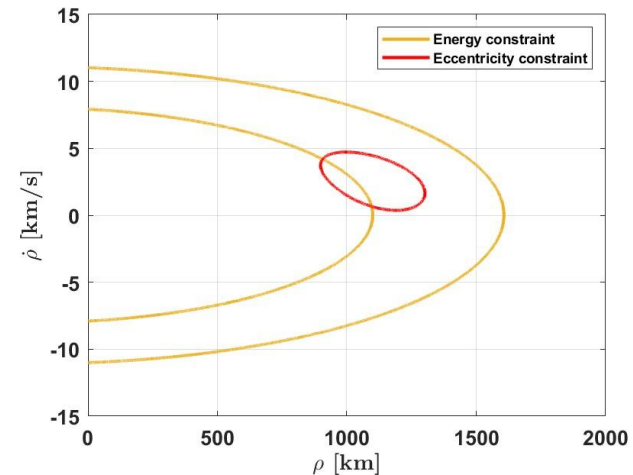
$$\mathbf{r} = \mathbf{q} + \rho \hat{\mathbf{p}}$$
$$\dot{\mathbf{r}} = \dot{\mathbf{q}} + \dot{\rho} \hat{\mathbf{p}} + \rho \frac{d\hat{\mathbf{p}}}{dt}$$

Angular track compression

Optical admissible region

Add radar constraints

Orbital states candidates



A. Milani, et al., *Orbit determination with very short arcs. I admissible regions*, *Celestial Mechanics and Dynamical Astronomy*, 2004
M.F. Montaruli, *Radar initial orbit determination method from angular track and Doppler shift measurements*, *Astrodynamics*, 2025



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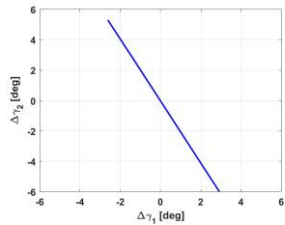


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IOD angles and DS



$$\mathbf{r} = \mathbf{q} + \rho \hat{\mathbf{p}}$$

$$\dot{\mathbf{r}} = \dot{\mathbf{q}} + \dot{\rho} \hat{\mathbf{p}} + \rho \frac{d\hat{\mathbf{p}}}{dt}$$

Angular track compression

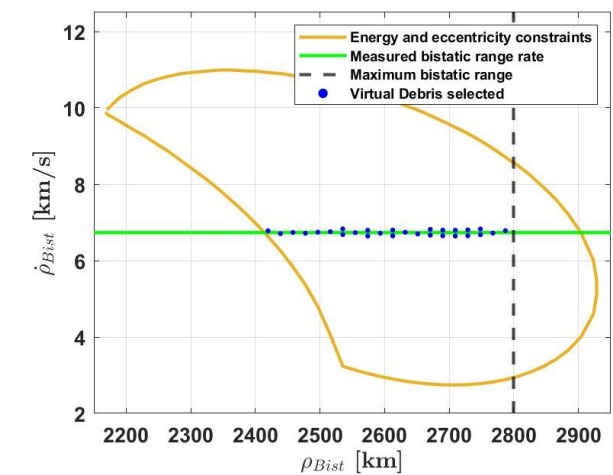
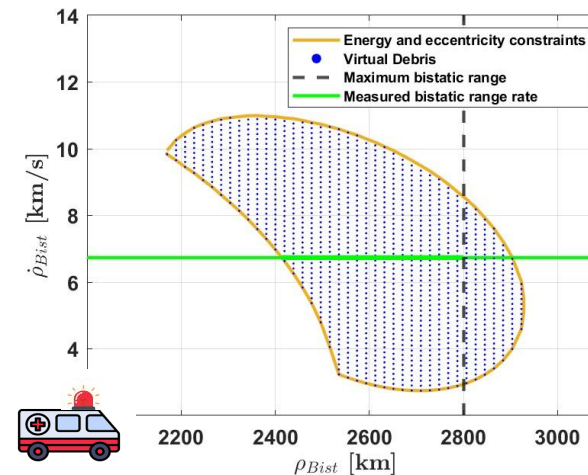
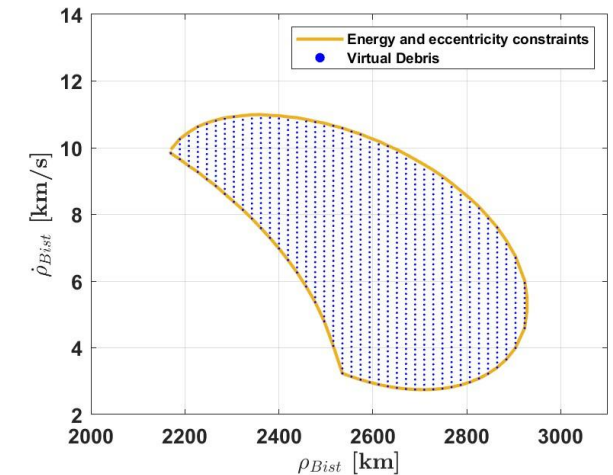
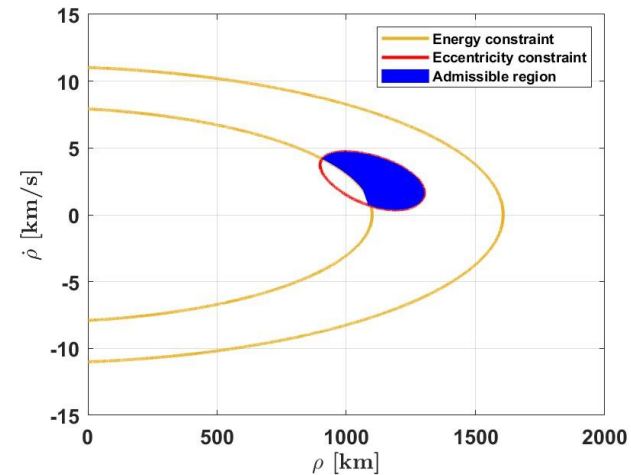
Optical admissible region

Bistatic case

Project VDs in the bistatic plane

Add radar constraints

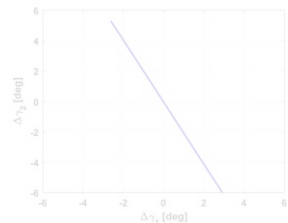
Orbital states candidates



A. Milani, et al., Orbit determination with very short arcs. I admissible regions, *Celestial Mechanics and Dynamical Astronomy*, 2004
M.F. Montaruli, Radar initial orbit determination method from angular track and Doppler shift measurements, *Astrodynamic*s, 2025

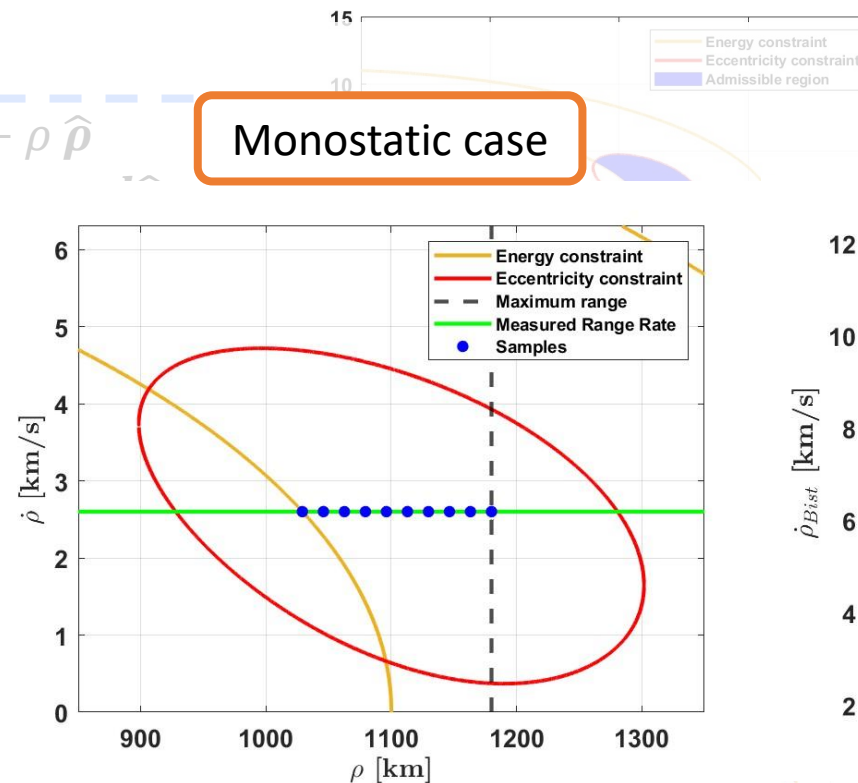


IOD angles and DS

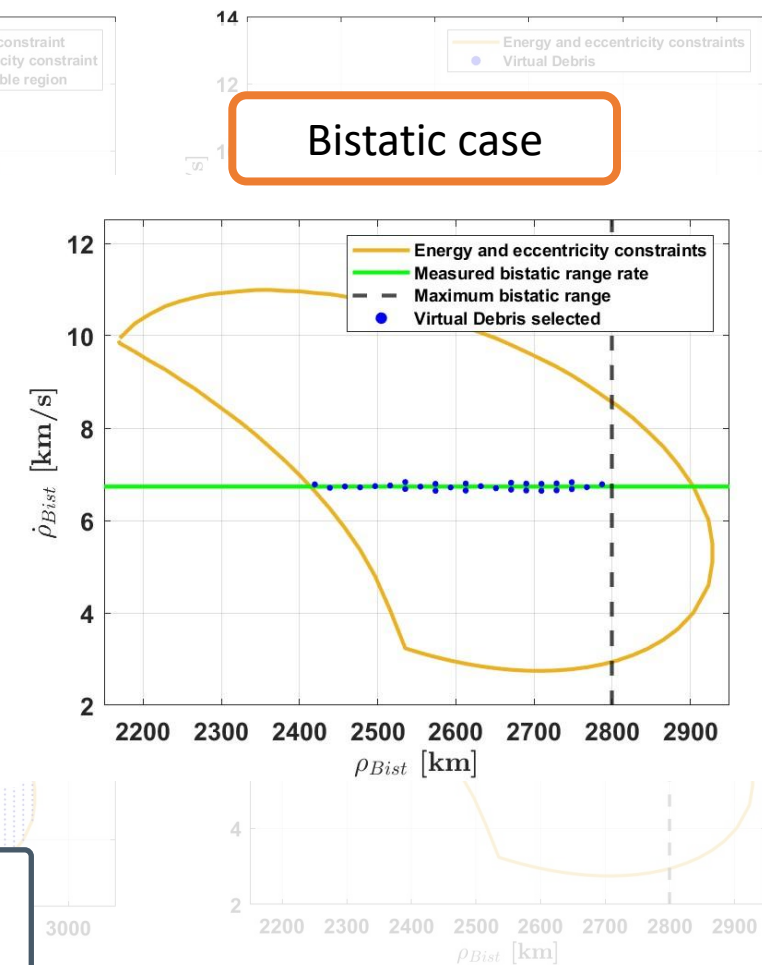


$$\mathbf{r} = \mathbf{q} + \rho \hat{\mathbf{p}}$$
$$\dot{\mathbf{r}} = \dot{\mathbf{q}} + \dot{\rho} \hat{\mathbf{p}}$$

Monostatic case



Bistatic case



Angular track compression

Optical admissible region

Bistatic case

Project VDs in the bistatic plane

Add radar constraints

Orbital state

- Select the candidate with the minimum eccentricity
- Run a batch filter

A. Milani, et al., Orbit determination with very short arcs. I admissible regions, *Celestial Mechanics and Dynamical Astronomy*, 2004

M.F. Montaruli, Radar initial orbit determination method from angular track and Doppler shift measurements, *Astrodynamic*, 2025



IOD angles and DS

Measurements accuracy

- Angles: $5e-02 \div 5e-01$ [deg]
- Range rate: $1.4e-02 \div 3.2e-02$ [km/s]

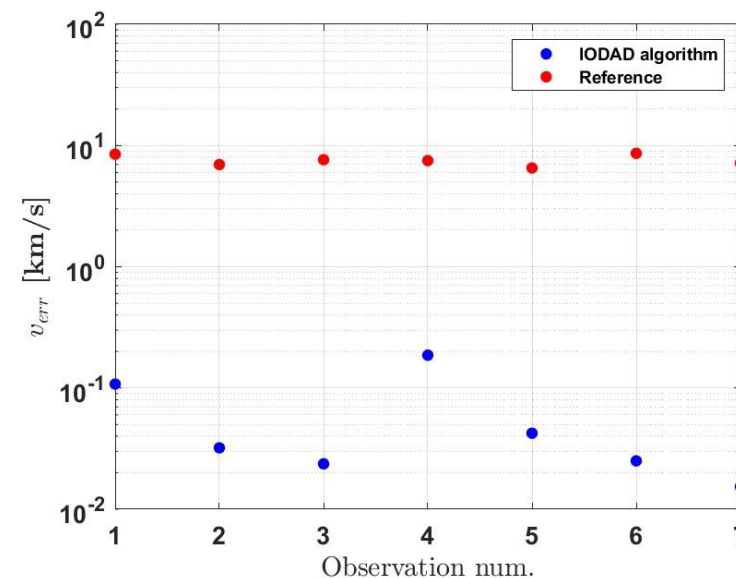
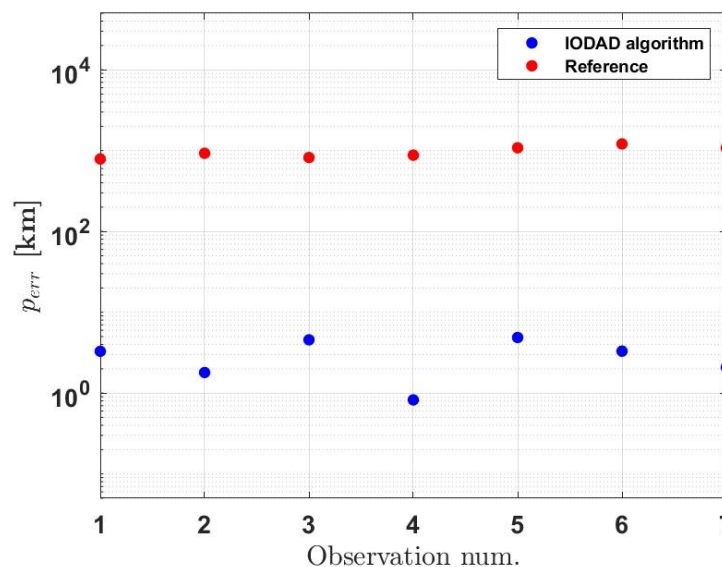
Duration: 20 s $\div$ 45 s

IODAD errors:

- Position: $8.2e-01 \div 4.9e+00$ km
- Velocity: $1.5e-02 \div 1.9e-01$ km/s

Reference methodology errors^[1]:

- Position: $7.9e+02 \div 1.2e+03$ km
- Velocity: $6.5e+00 \div 8.6e+00$ km/s



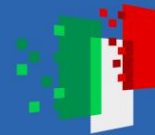
- ✓ Remarkably improvement with respect the state of the art
- ✓ Cataloguing capability from UCTs without range



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OUTLINE

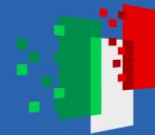
- 01 MATER algorithm
- 02 Operational implementation
- 03 Ambiguity solving through East-West branch
- 04 Radar IOD
- 05 Fragmentation campaigns (CZ-6A, Starlink 34343, ZQ-2E R/B)
- 06 Conclusions



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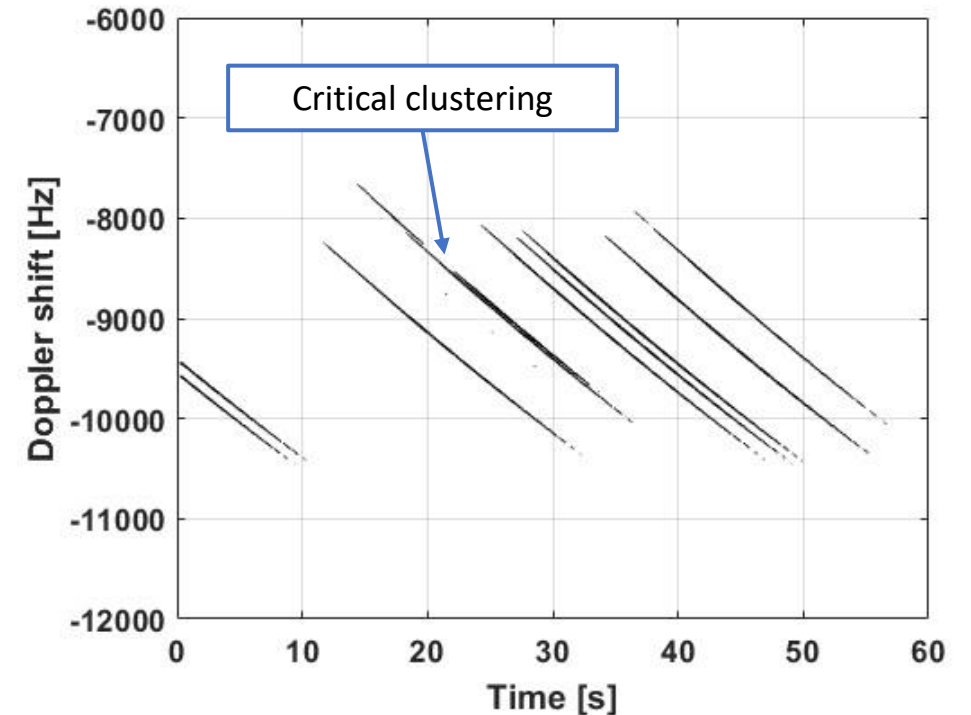


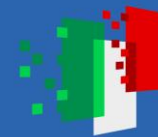
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Fragmentation CZ-6A

- September 3°, 2024
- Target: CZ-6A DEB (norad id 60649)

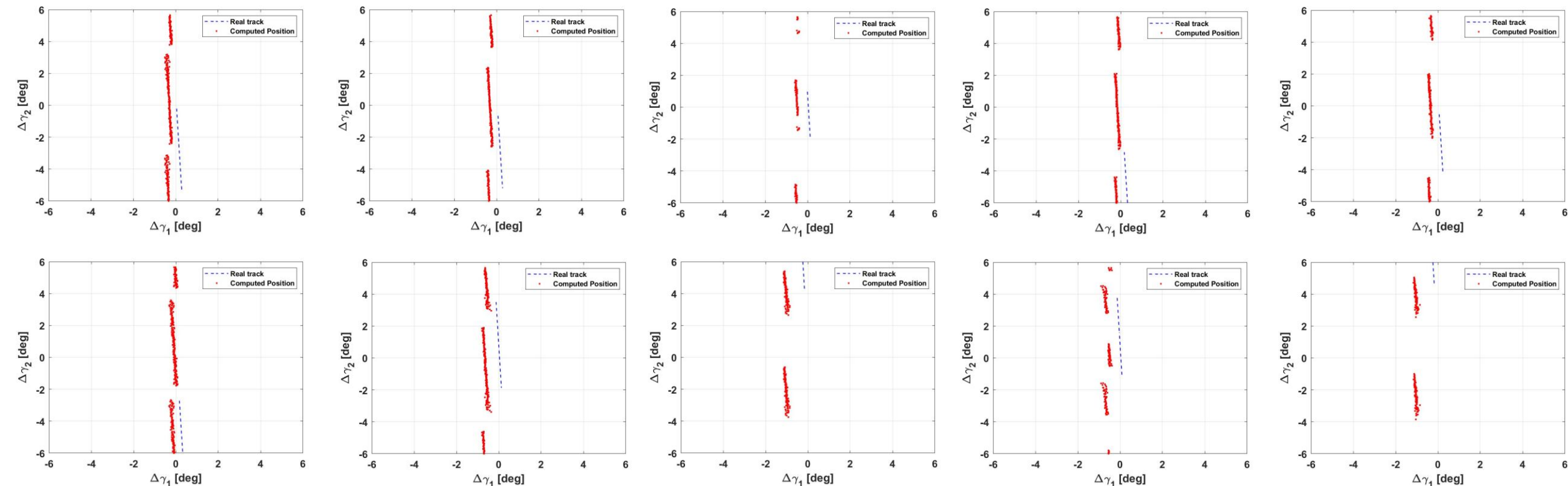
- August 7°, 2024: break-up of the launcher carrying QIANFAN constellation satellites
- QIANFAN constellation detected by 1N and 2N
- Signal transmitted: 60% CHIRP + 40% CW





Fragmentation CZ-6A

- Red dots: signal direction of arrival detected.
- Dashed blue line: target prediction.





Fragmentation CZ-6A

2 fragments detected

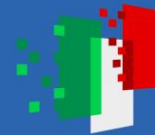
		TDM A	TDM B	TDM C	TDM D	TDM E	TDM F	TDM G	TDM H	TDM I	TDM L
QIANFAN-1	60379							0.38			
QIANFAN-5	60383						0.31				
QIANFAN-6	60384				0.35	10.35					
QIANFAN-7	60385			0.84						1.06	
QIANFAN-8	60386	1.36	0.49								
QIANFAN-9	60387	15.53	5.59			0.36					
QIANFAN-11	60389			0.29					0.29	0.5	
QIANFAN-12	60390								0.37		4.73
QIANFAN-13	60391								2.78		0.34
QIANFAN-14	60392					5.58					
CZ-6A DEB	60649			13.61						12.23	
CZ-6A DEB	60800										18.9



Fragmentation CZ-6A

We keep one track per target and one track per target

		TDM A	TDM B	TDM C	TDM D	TDM E	TDM F	TDM G	TDM H	TDM I	TDM L
QIANFAN-1	60379							0.38			
QIANFAN-5	60383						0.31				
QIANFAN-6	60384				0.35	10.35					
QIANFAN-7	60385			0.84						1.06	
QIANFAN-8	60386	1.36	0.49								
QIANFAN-9	60387	15.53	5.59			0.36					
QIANFAN-11	60389			0.29					0.29	0.5	
QIANFAN-12	60390								0.37		4.73
QIANFAN-13	60391								2.78		0.34
QIANFAN-14	60392					5.58					
CZ-6A DEB	60649			13.61						12.23	
CZ-6A DEB	60800										18.9



Fragmentation CZ-6A

TDM A remains not correlated. Possible fragment?

		TDM A	TDM B	TDM C	TDM D	TDM E	TDM F	TDM G	TDM H	TDM I	TDM L
QIANFAN-1	60379							0.38			
QIANFAN-5	60383						0.31				
QIANFAN-6	60384				0.35						
QIANFAN-7	60385									1.06	
QIANFAN-8	60386		0.49								
QIANFAN-9	60387					0.36					
QIANFAN-11	60389			0.29							
QIANFAN-12	60390								0.37		
QIANFAN-13	60391										0.34
QIANFAN-14	60392										
CZ-6A DEB	60649										
CZ-6A DEB	60800										



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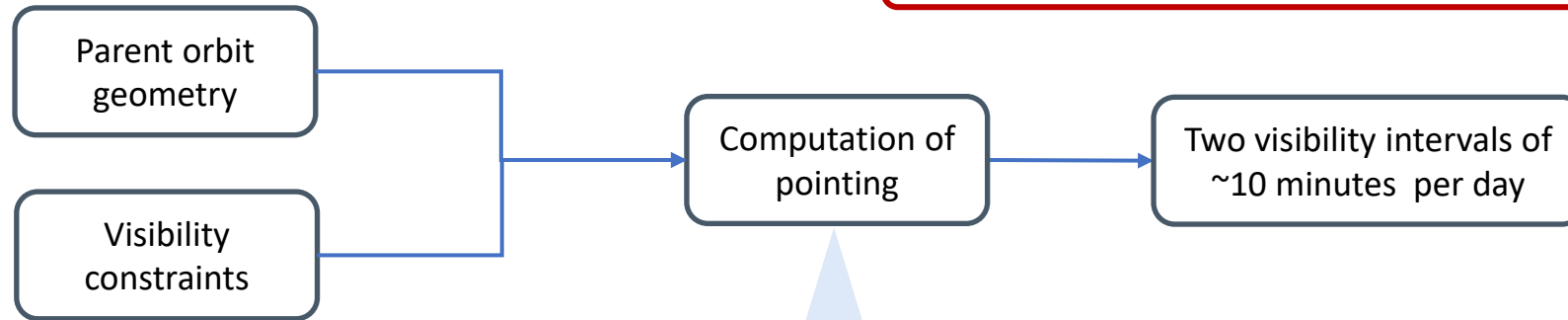
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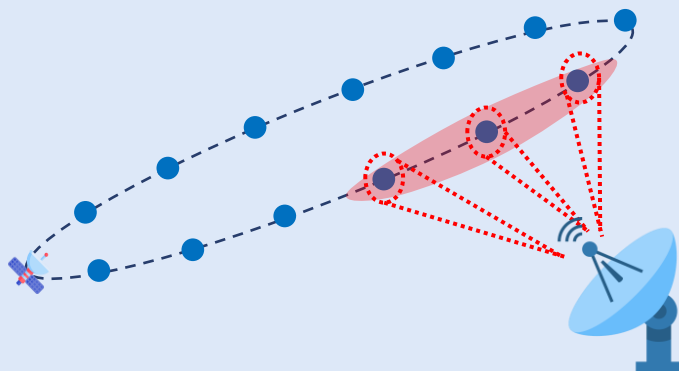
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Fragmentation Starlink 34343

Fragmentation epoch: around 2026-03-29T19:30:00.000



Orbital plane scanning



+

Meridian crossing





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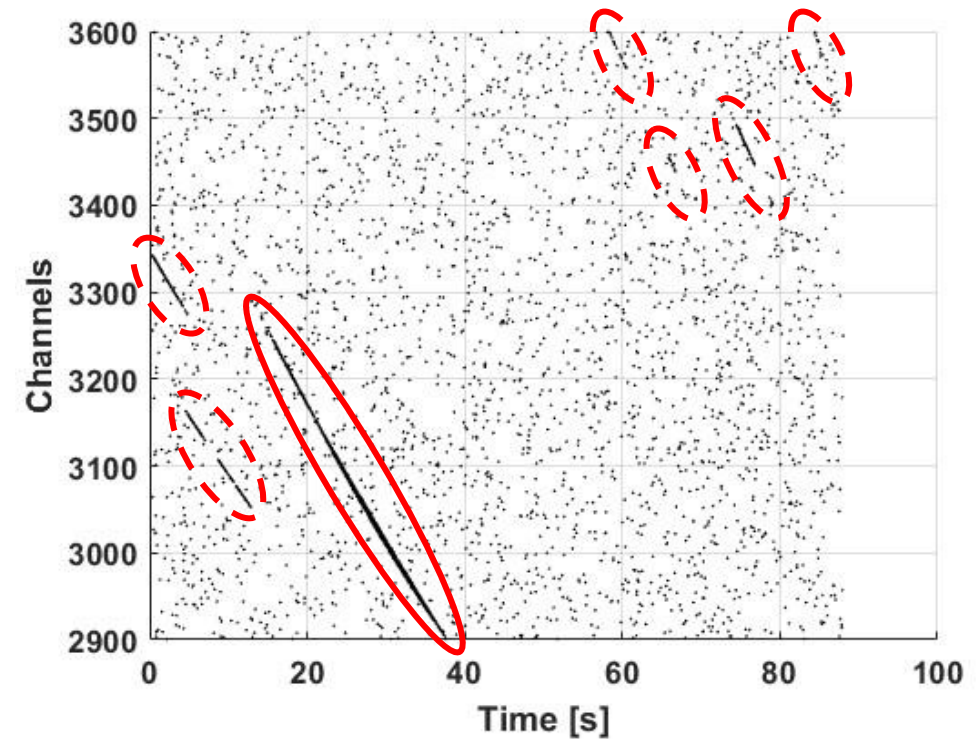
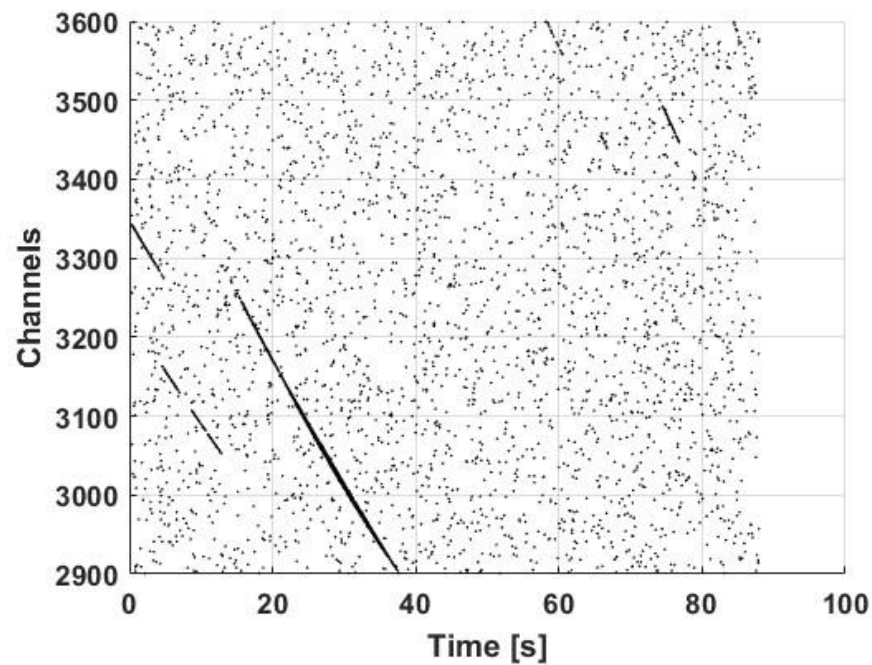


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Fragmentation Starlink 34343

Fragmentation epoch: around 2026-03-29T19:30:00.000

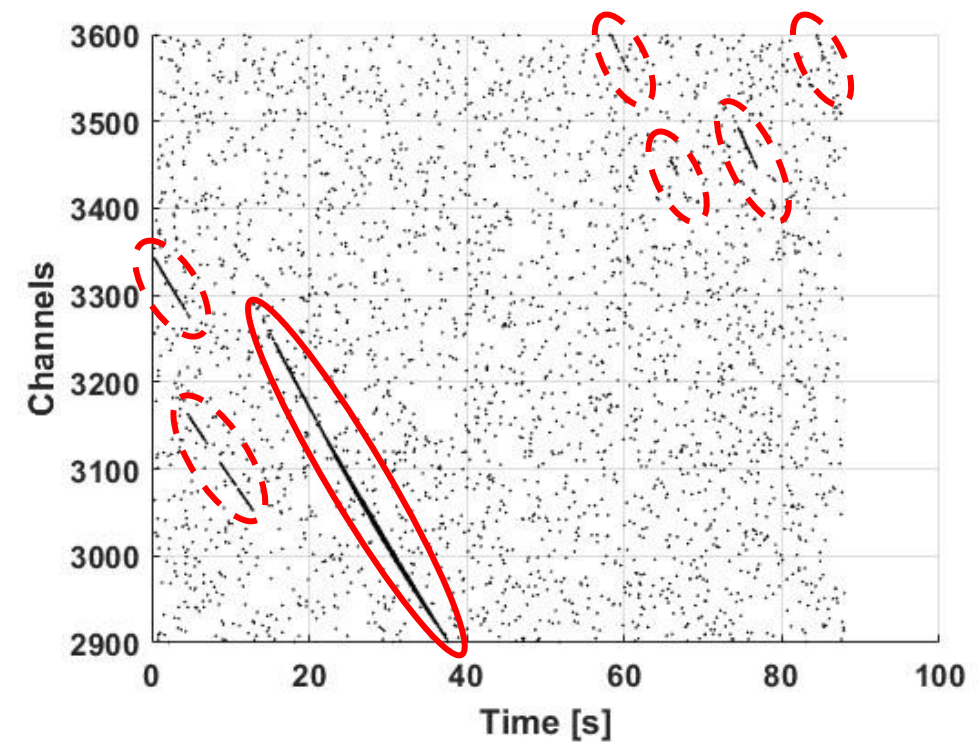
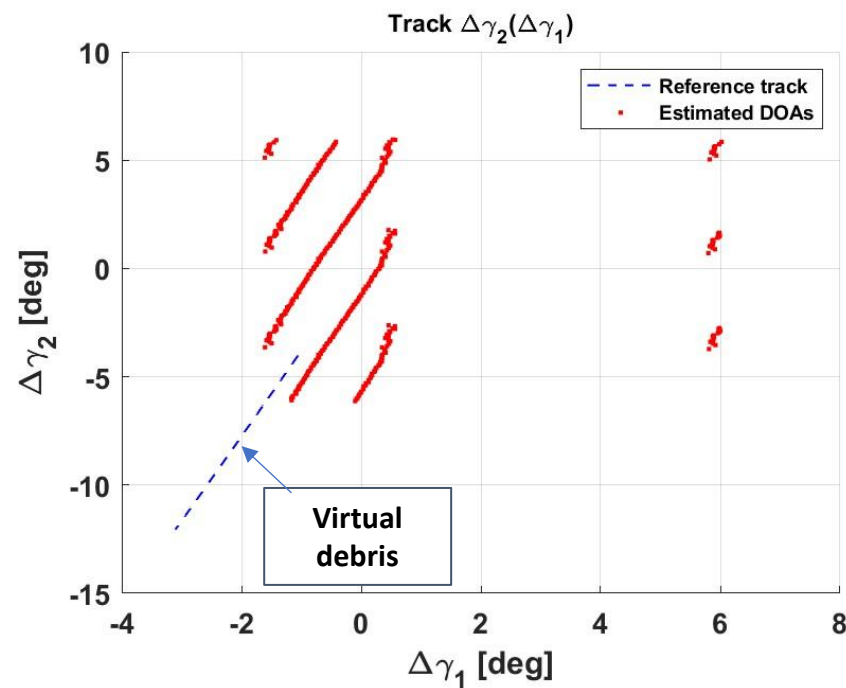
BIRALES detection from 02/04/2026 – h. 19:38 UTC





Fragmentation Starlink 34343

BIRALES detection from 02/04/2026 – h. 19:38 UTC





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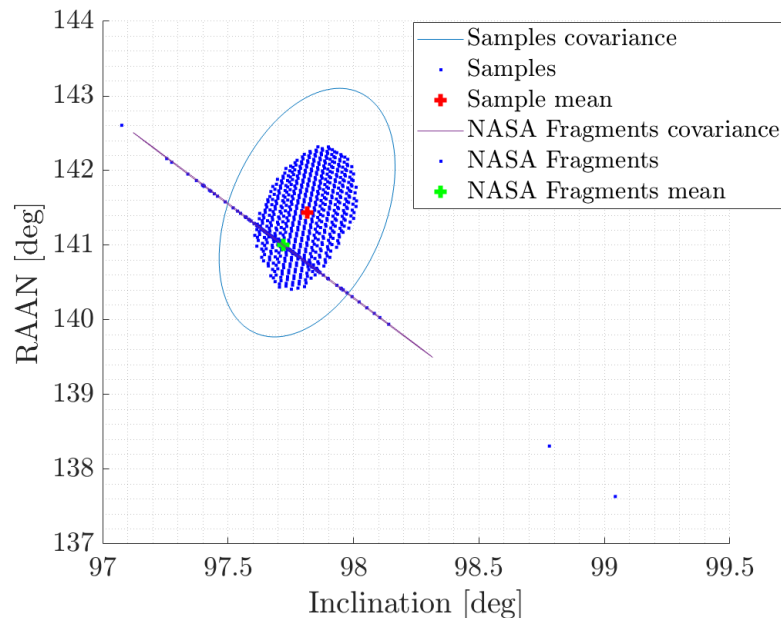


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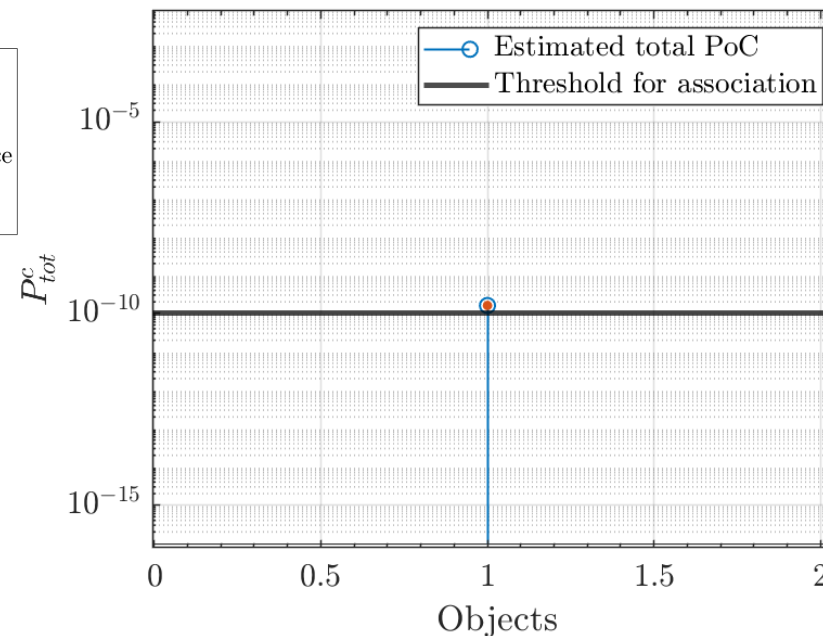
Fragmentation Starlink 34343

Fragmentation forensic analysis

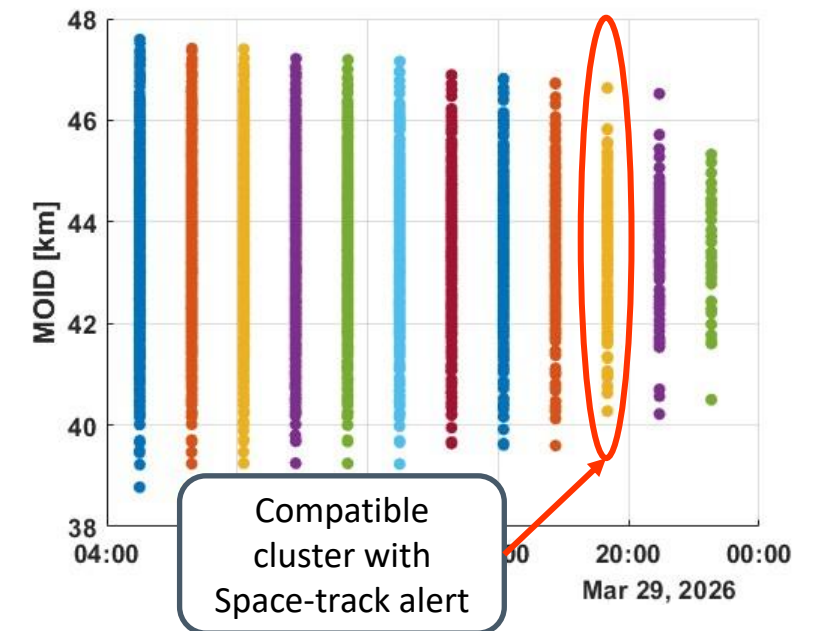
Measurement space
(no IOD result exploited)



Orbital space
(based on IOD result)



FRED: Fragmentation Epoch
Detector



M.F. Montaruli, et al., A stochastic approach to detect fragmentation epoch from a single fragment orbit determination, *Advances in Space Research*, 2023

Space-track website: <https://www.space-track.org/>

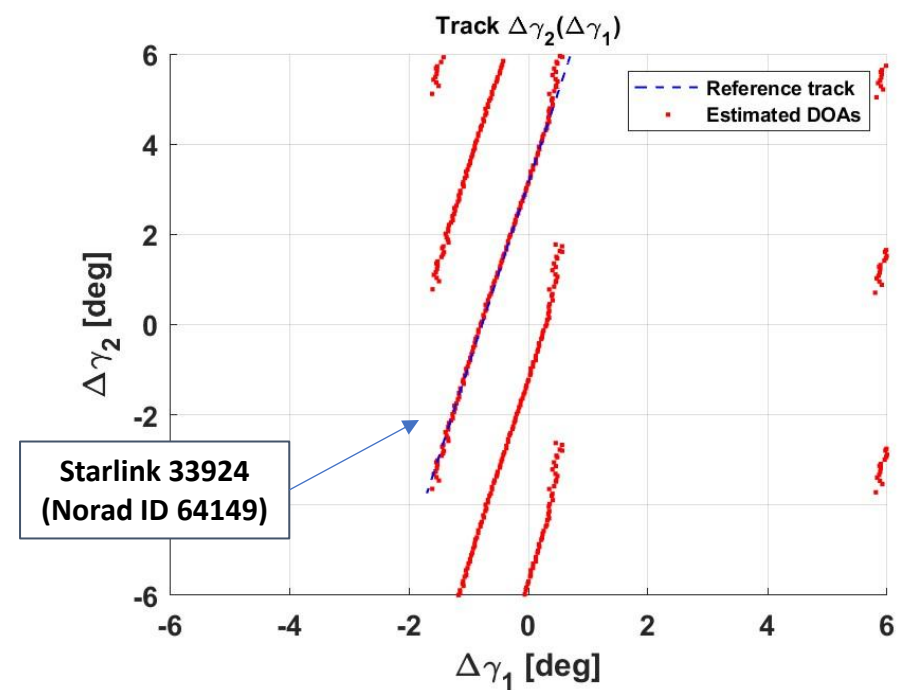
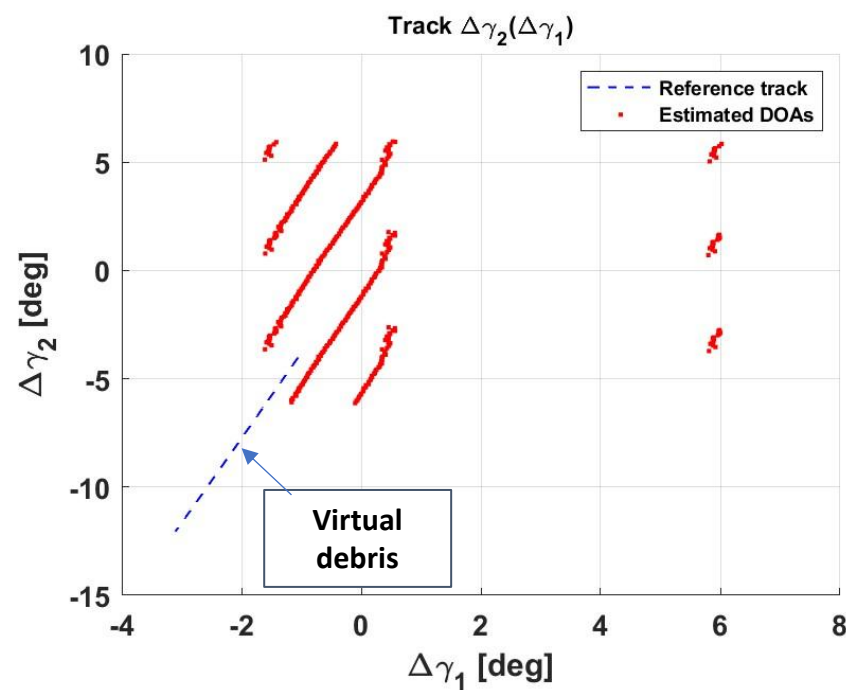
P. Grattagliano, et al. Early stage characterization of on-orbit fragmentation events, *Acta Astronautica*, 2025

P. Grattagliano, et al., Advanced techniques for the stochastic fragmentation event characterization from a single fragment orbital state, *Advances in Space Research*, 2026



Fragmentation Starlink 34343

BIRALES detection from 02/04/2026 – h. 19:38 UTC





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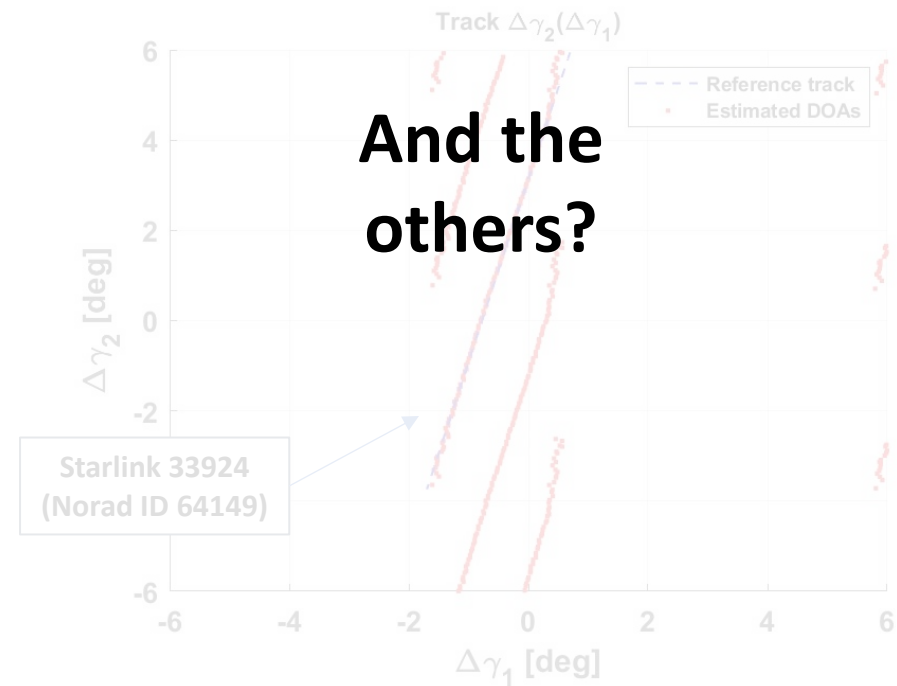
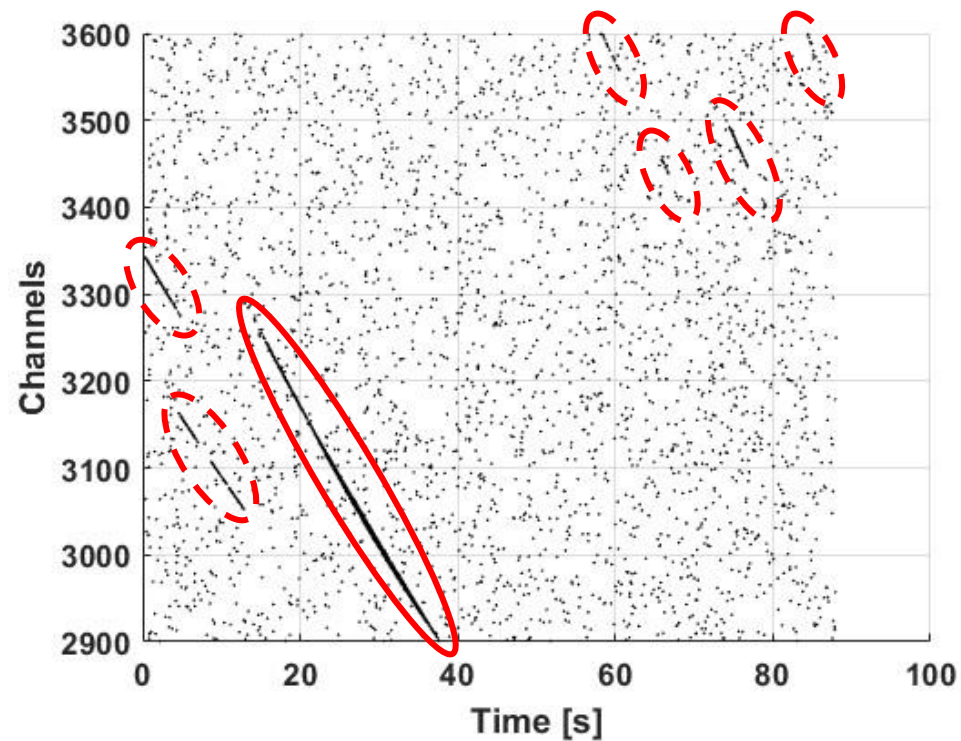
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Fragmentation Starlink 34343

BIRALES detection from 02/04/2026 – h. 19:38 UTC

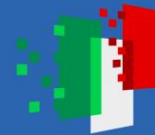




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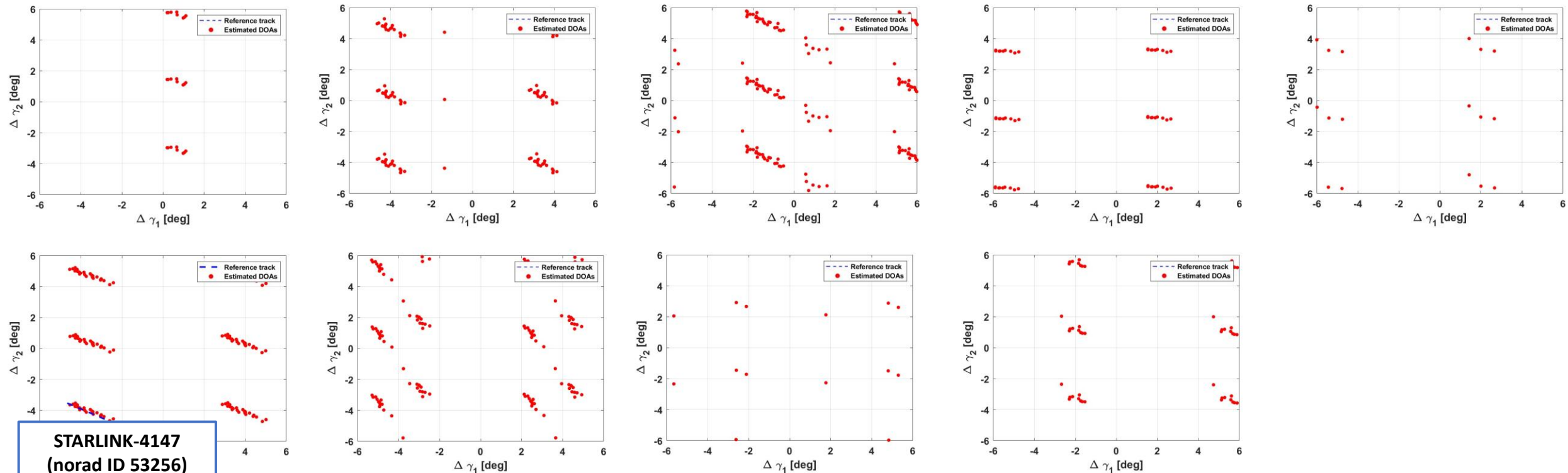


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Fragmentation Starlink 34343



- Uncatalogued, but not associated to the events.
- Starlinks not yet in the catalogue?



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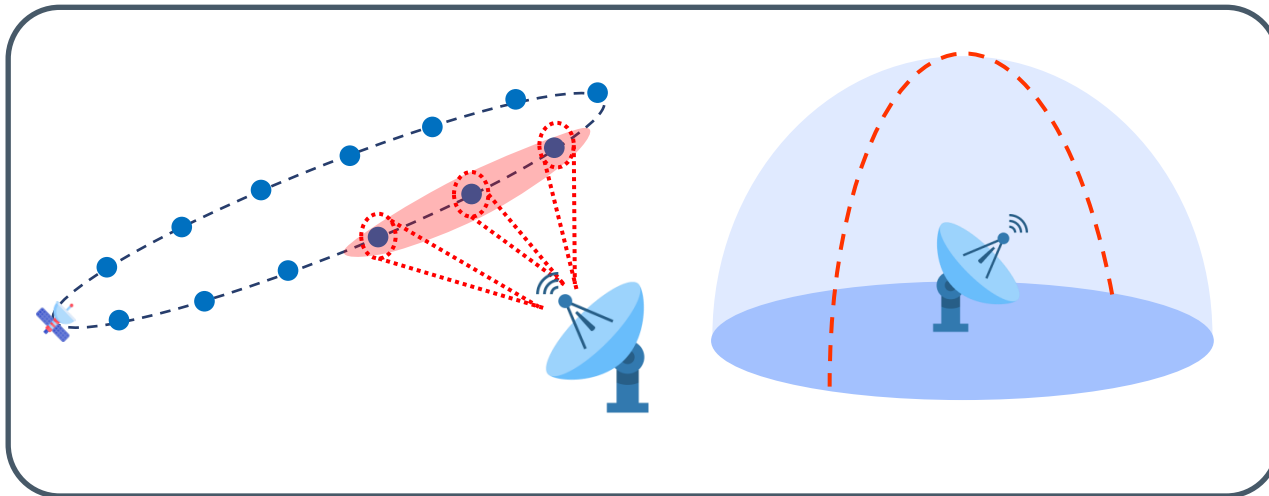


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Fragmentation ZQ-2E R/B

Fragmentation epoch: around 2026-06-09T08:47:00.000



2026-06-20T03:33:31.603	+/- 5 minutes	✓
2026-06-20T10:00:07.230	+/- 5 minutes	
2026-06-21T09:38:03.110	+/- 5 minutes	
2026-06-22T02:36:11.689	+/- 5 minutes	
2026-06-22T09:08:09.316	+/- 5 minutes	
2026-06-23T02:08:42.894	+/- 5 minutes	✓
2026-06-23T08:34:03.196	+/- 5 minutes	



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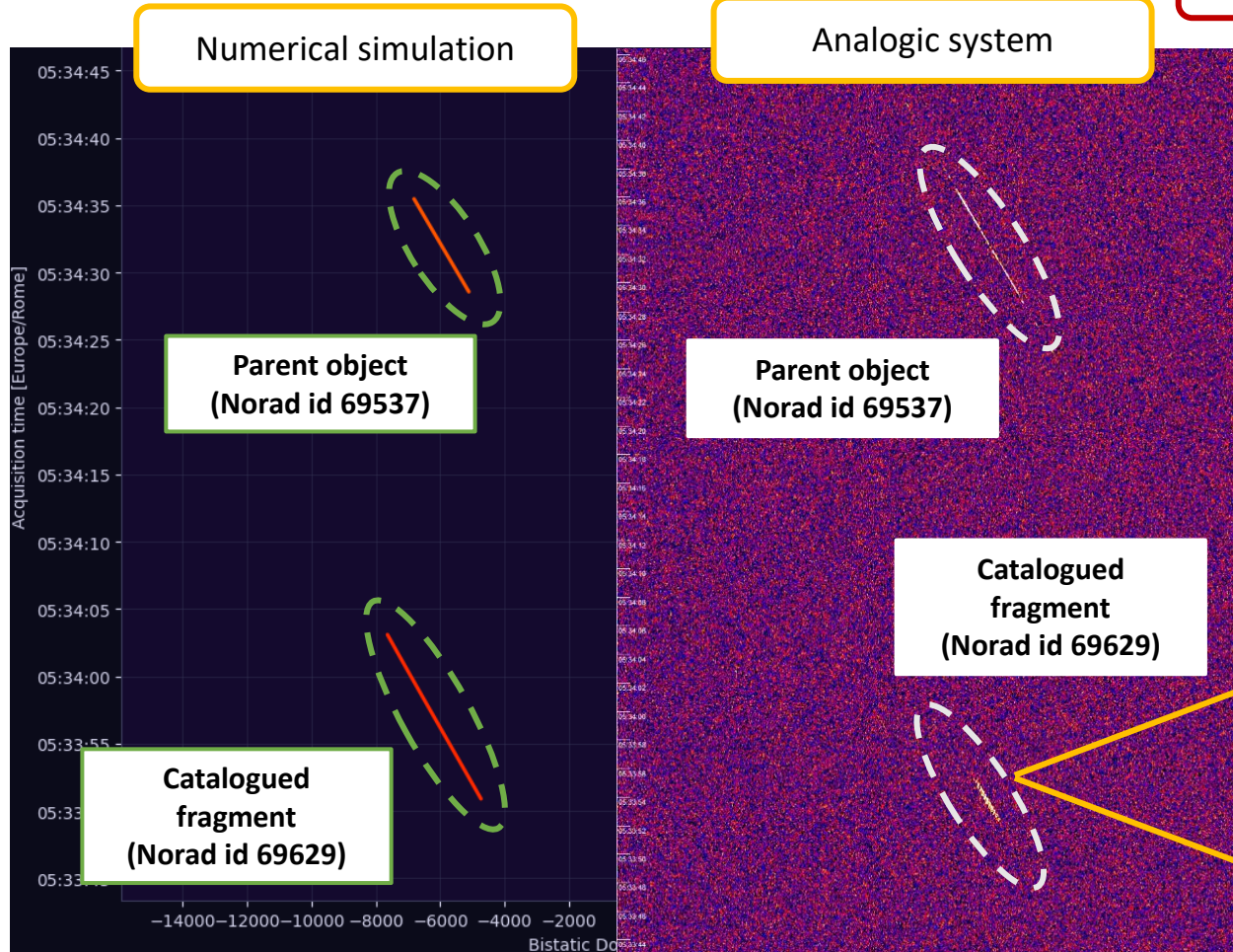
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Fragmentation ZQ-2E R/B

Fragmentation epoch: around 2026-06-09T08:47:00.000



- Detection of both parent object (the target) and a catalogued fragment.
- The old system correlated the catalogued fragment to a Starlink-31342 (Norad Id 59166).
- Validation of the observation strategy.
- The orbital plane based strategy remains robust for several days.
- Is the fragment tumbling?

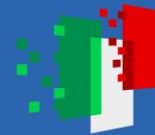
Importance of migrating to the
new operational system



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OUTLINE

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Conclusions

- ✓ MATER algorithm operationally implemented
- ✓ Potential of E-W branch contribution to solve the angular track ambiguity
 - No ambiguous tracks for uncorrelated tracks
- ✓ Radar Initial Orbit Determination from angles and Doppler shift
 - Capability to determine the target orbital state even if the range is not present
- ✓ Effectiveness in fragmentation campaigns
 - Dedicated observation strategy validated
 - Criticalities posed by mega-constellations
 - Uncorrelated tracks detected



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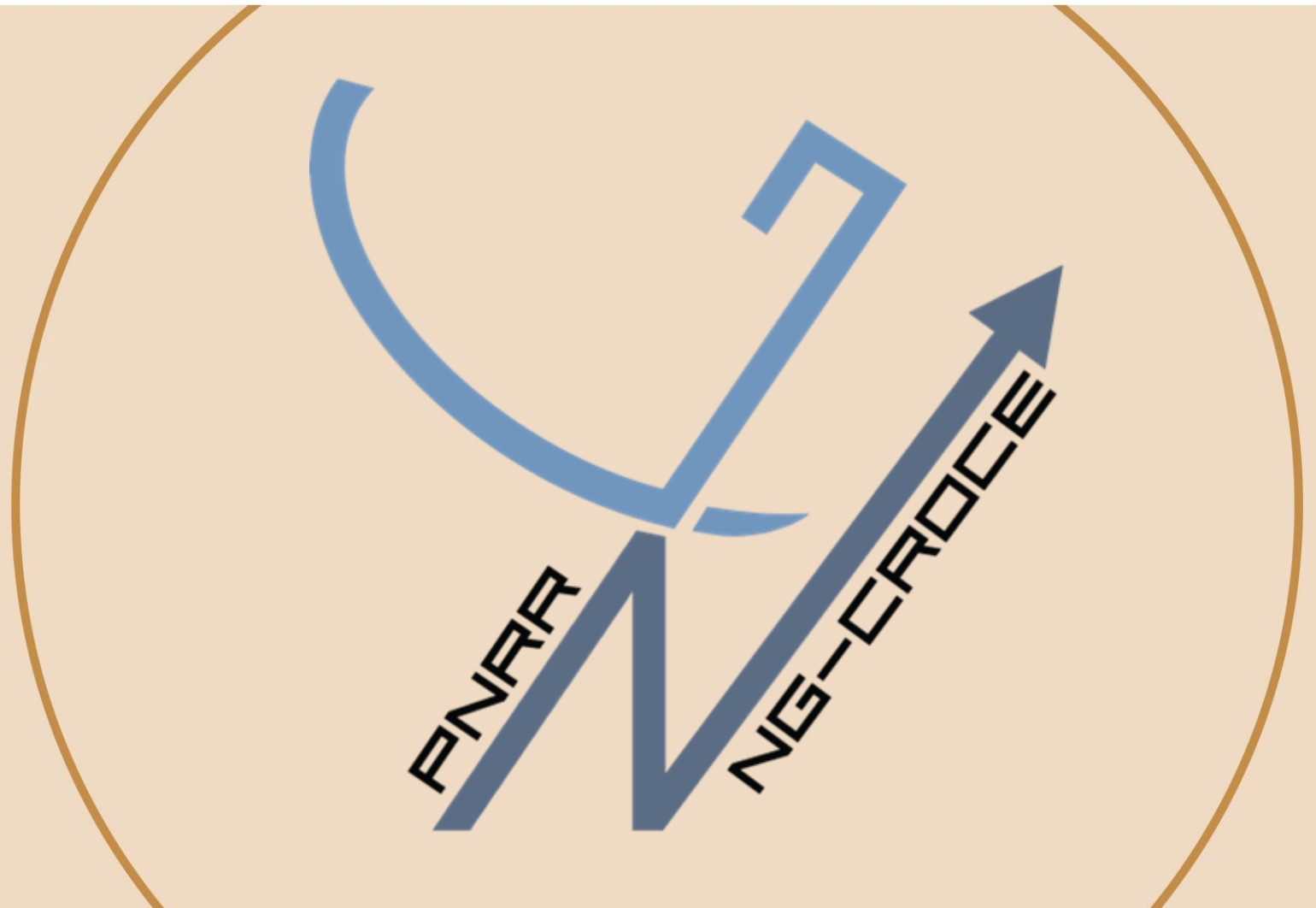
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Thank you

Final Meeting

Martedì 30 Giugno – Giovedì 02 Luglio

OAC - Osservatorio Astrofisico di Arcetri



M.F. Montaruli