

LGWA Tilt-Platform Design and Performance Constraints

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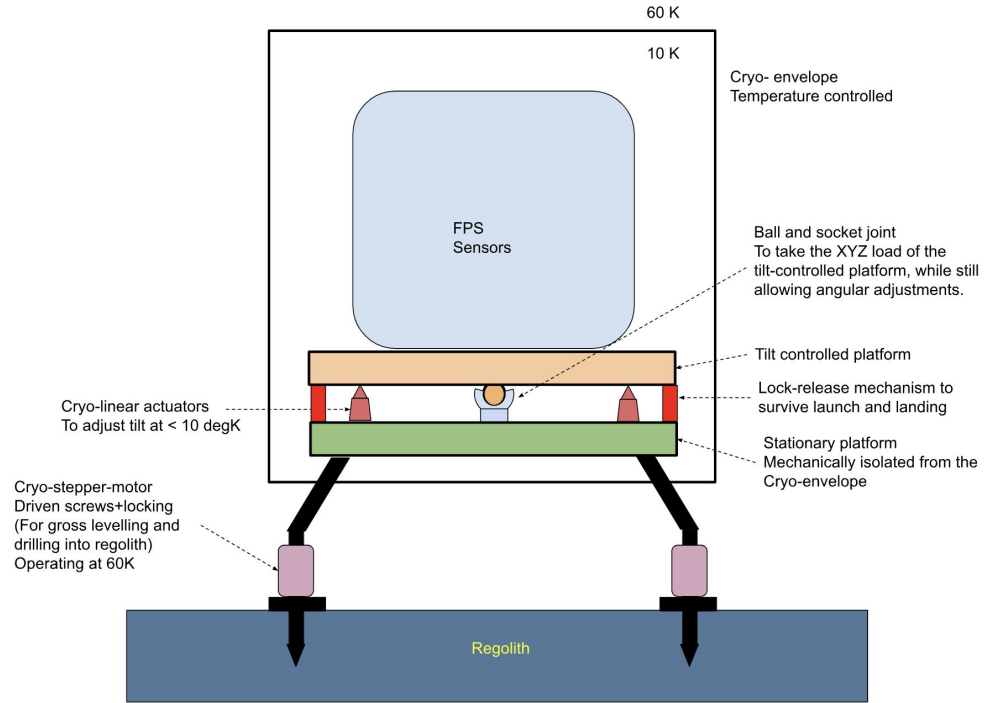


Basic Considerations

- Payload requirements
 - Lunar operation : cryogenic operation, structural rigidity,
 - Lab tests on Earth : thermal cycling, Weight on earth
- Mission requirements
 - Power consumption, Radiation hardening, redundancy/failure of mission critical components, weight limitations,
- Launch requirements
 - Vibration load, shock load, locking mechanism
- Landing requirements:
 - Landing shock, Deployment strategy, Unlocking mechanism
- Operation requirements:
 - Alignment precision, In-situ calibration, health checks, lunar quakes, regolith settling, data-handling and transmission, long term operation,

Basic Structure

- Two levels of actuators
 - Coarse control - on regolith, 50-100 K supporting whole platform + sensors
 - Fine control - minimal load, < 10 K
 - Independent of locking/unlocking mechanism
- Potential cryo-linear actuators
 - JPE : CLA2602
 - To be tested
 - Not yet-space qualified
- Fine Control actuators
 - do not need to support load of the sensors / platform
 - Ball and socket joint to take the load and provide XYZ, θ_z constraints
 - Not in contact during launch / landing



Ongoing work

- Payload team is working on a detailed specifications list.
- Requires constraints from full mission parameters
- Assessment of technological readiness of various components
- Proposals for development of testing platforms and tilt sensors underway

Platform Levelling System Specifications for LGWA

Operating Temperature:	15 K - 100 K	<i>(operation during testing at room T?)</i>
Thermal cycling		
15 K to Room T		100 cycles
15 k to 100 K		1000 cycles
Life cycle		
roundtrips of 60 mm		50
roundtrips of 5 mm		100
roundtrips of 0.1 mm		10000
Payload mass on the platform:	30 kg	
Levelling platform width:	50 cm	
Duration of the mission:	10 yrs	

Excerpt from the tilt-platform specs

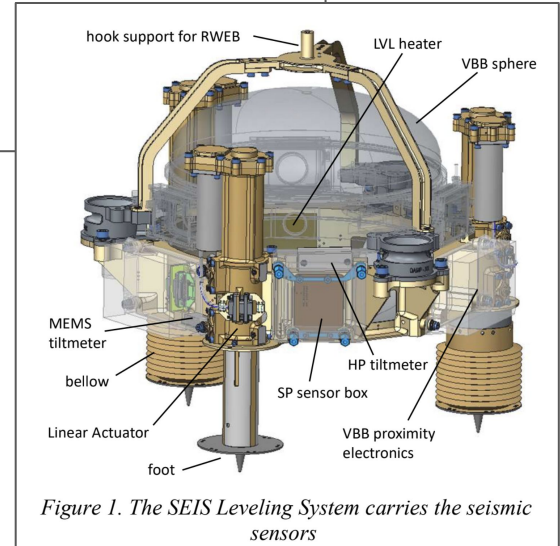


Figure 1. The SEIS Levelling System carries the seismic sensors

Plans for the near future

- Requires a joint study including the space technology community
- Requires planning in conjunction with the Strategic Mission Plan
- Requires demonstration of all technologies till TR6 level within 3 years
 - Planned testing activity at IUCAA
(in collab with NPL, UniCam, GSSI)
 - Planned testing facilities at GSSI



Lessons from Mars Insight Mission

- Tested technologies
- concepts we can borrow

<https://x.com/NASAINsight/status/1571892419624599552>



Questions / Remarks / Comments

Welcome!

Thank You!